

# NANOPHASE PRECIPITATION STRENGTHENED AL ALLOYS PROCESSED THROUGH THE AMORPHOUS STATE

**Publication number:** WO03104505

**Publication date:** 2003-12-18

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**Classification:**

- **International:** B22D23/00; C22C1/00; C22C21/00; C22C45/08; C22F1/00; C22F1/04; B22D23/00; C22C1/00; C22C21/00; C22C45/00; C22F1/00; C22F1/04; (IPC1-7): C22C

- **European:** C22C21/00; C22C45/08; C22F1/04

**Application number:** WO2003US12622 20030424

**Priority number(s):** US20020375940P 20020424; US20030450114P 20030225

**Also published as:**

- [PA] WO03104505 (A3)
- [EP] EP1499753 (A3)
- [EP] EP1499753 (A2)
- [EP] EP1499753 (A0)
- [AU] AU2003265234 (A1)

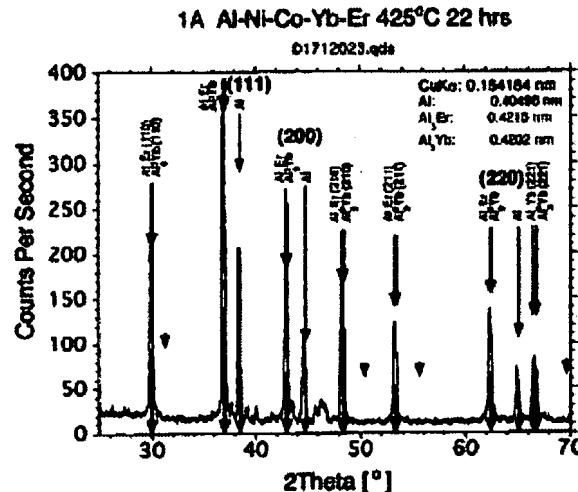
**Cited documents:**

- [DE] US5397403
- [DE] XP008025999
- [DE] XP001172942
- [DE] XP008025988
- [DE] XP008025998

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## Abstract of WO03104505

Aluminum alloys having improved strength characteristics at elevated temperatures (300 DEG C) are manufactured by combining selected transition metals (Ni, Co, Ti, Fe, Y, Sc) and selected rare earth materials (Er, Tm, Yb, Lu) in amounts of about 2 to 12% and 2 to 15% atomic percent respectively in an amorphous, glassy state and subsequently devitrifying the amorphous material to form a crystalline mix of fcc and L12 phase material. Devitrification from the amorphous state may be effected by various means including thermal and thermo mechanical processes.



Data supplied from the esp@cenet database - Worldwide

(19) World Intellectual Property Organization  
International Bureau(43) International Publication Date  
18 December 2003 (18.12.2003)

PCT

(10) International Publication Number  
**WO 03/104505 A2**(51) International Patent Classification<sup>7</sup>:**C22C**

(21) International Application Number:

PCT/US03/12622

(22) International Filing Date:

24 April 2003 (24.04.2003)

(25) Filing Language:

English

(26) Publication Language:

English

(30) Priority Data:

60/375,940 24 April 2002 (24.04.2002) US  
60/450,114 25 February 2003 (25.02.2003) US(71) Applicant (for all designated States except US):  
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(81) Designated States (national): AE, AG, AL, AM, AT, AU, AZ, BA, BB, BG, BR, BY, BZ, CA, CH, CN, CO, CR, CU, CZ, DE, DK, DM, DZ, EC, EE, ES, FI, GB, GD, GE, GH, GM, HR, HU, ID, IL, IN, IS, JP, KE, KG, KP, KR, KZ, LC, LK, LR, LS, LT, LU, LV, MA, MD, MG, MK, MN, MW, MX, MZ, NI, NO, NZ, OM, PH, PL, PT, RO, RU, SC, SD, SE, SG, SK, SL, TJ, TM, TN, TR, TT, TZ, UA, UG, US, UZ, VC, VN, YU, ZA, ZM, ZW.

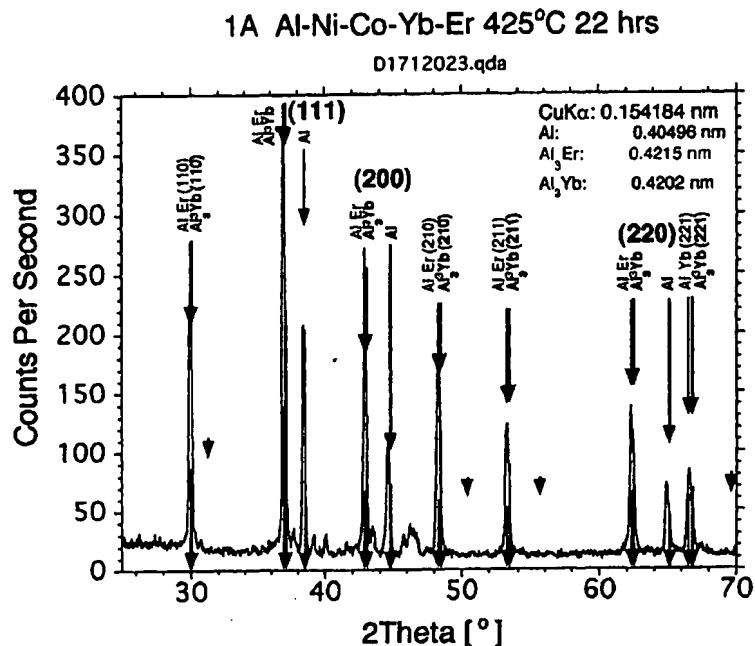
(84) Designated States (regional): ARIPO patent (GH, GM, KE, LS, MW, MZ, SD, SL, SZ, TZ, UG, ZM, ZW), Eurasian patent (AM, AZ, BY, KG, KZ, MD, RU, TJ, TM), European patent (AT, BE, BG, CH, CY, CZ, DE, DK, EE, ES, FI, FR, GB, GR, HU, IE, IT, LU, MC, NL, PT, RO, SE, SI, SK, TR), OAPI patent (BF, BJ, CF, CG, CI, CM, GA, GN, GQ, GW, ML, MR, NE, SN, TD, TG).

## Published:

— without international search report and to be republished upon receipt of that report

[Continued on next page]

(54) Title: NANOPHASE PRECIPITATION STRENGTHENED AL ALLOYS PROCESSED THROUGH THE AMORPHOUS STATE





*For two-letter codes and other abbreviations, refer to the "Guidance Notes on Codes and Abbreviations" appearing at the beginning of each regular issue of the PCT Gazette.*

## NANOPHASE PRECIPITATION STRENGTHENED AI ALLOYS PROCESSED THROUGH THE AMORPHOUS STATE

### BACKGROUND OF THE INVENTION

5 Activities relating to the development of the subject matter of this invention were funded at least in part by United States Government, U.S. Army Aviation & Missile Command Contract No. DAAH01-02-C-R125, and thus may be subject to license rights and other rights in the United States.

10 In a principal aspect, the present invention relates to aluminum-based alloys processed through an amorphous state or stage, preferably by means of a rapid solidification process from molten alloy, and then devitrified to a primarily crystalline structure by thermal or thermomechanical processing. In order to achieve a desired microstructure during processing, the aluminum alloy comprises a combination of rare earth and transition metal components with the aluminum. The final crystalline microstructure has stable high temperature strength (at 15 or above about 300°C) characterized by nanoscale intermetallic precipitates, preferably L1<sub>2</sub> phase precipitates.

20 Currently available commercial aluminum alloys, either manufactured with ingot or powder processing, are not capable of simultaneously achieving high strength and high temperature stability near 300°C; such characteristics being particularly important in applications such as the fan components in turbine engines where short-term tensile strengths of the order of 500MPa (40-60 ksi) are desired. Precipitation hardening introduced by aging is a known method to strengthen aluminum alloys. Thus, conventional high strength aluminum alloys in commercial applications utilize GP (Guinier-Preston) zones and subsequent precipitation when seeking to achieve high material strength. With this type of strengthening, 25 precipitates are usually formed during aging at or below 250°C in order to produce appropriate strengthening precipitate dispersions.

30 Examples of aluminum alloys processed with relatively high aging temperatures in commercial practice include 2618 (200°C for 20 hours), 4032 (170-175°C for 8-12 hours), and 2218 (240°C for 6 hours) [Metals Handbook — Properties and Selection: Nonferrous Alloys and Special-Purpose Materials, Volume 2, 10<sup>th</sup> Edition, ASM International]. At the noted aging temperatures, these alloys have improved microstructure stability relative to other commercial aluminum alloys. These aluminum alloys, when age-hardened, usually possess a room temperature yield strength of about 600Mpa (85 ksi). However, at temperatures approaching

300°C, the precipitation strengthening efficiency in these alloys quickly and significantly degrades as a result of precipitate coarsening and/or dissolution. Due to the unstable strengthening precipitate size distribution at such high temperatures, the yield strength of currently available aluminum alloys at 300°C is often only 10% of the yield strength at room 5 temperature, and thus renders such alloys unsuitable for high temperature applications above 150°C.

In order to achieve a combination of high strength and usable high temperature properties in aluminum alloys, researchers have investigated a variety of intermetallic precipitation dispersions. Since selected intermetallic precipitates may determine the strength 10 and ductility of the final aluminum alloy, it is desirable to select cubic precipitate phases with high crystallographic symmetry to promote ductility. The aluminum-based L<sub>1</sub><sub>2</sub> phase (fcc ordered) is one of the best-known precipitates to achieve a good combination of high strength and high toughness.

However, since crystalline aluminum has very limited solubility for most intermetallics, 15 including L<sub>1</sub><sub>2</sub> forming components, it is difficult to produce fine intermetallic dispersions through crystalline solid-state heat treatments that utilize a step to dissolve the strengthening phase particles prior to reprecipitation during aging. However, rapid solidification processing (RSP) from the liquid state has been attempted or utilized. With an RSP process, it is possible to either (1) directly produce fine crystalline structure, or (2) produce partially or completely 20 amorphous aluminum alloys. Nonetheless, crystalline RSP aluminum alloys have not been able to meet the high temperature strength requirements due to the difficulty of producing small, stable particle sizes at adequate volume fraction. Alternatively, the focus on amorphous aluminum alloys has been primarily on the glass formability. These glassy alloys in general have low ductility, and, when devitrified, they typically form crystalline microstructures that 25 are far from optimal for fracture toughness and other requirements.

For example, RSP crystalline aluminum alloys, not designed to form glass with RSP, utilize a metastable and coherent L<sub>1</sub><sub>2</sub> dispersion (about 1%) which converts into other phase structures at high temperature. [Parameswaran, Weertman and Fine, Scripta Metall., vol.23, pp.147, 1989] U.S. Pat. Nos. 4,950,452, 5,074,935, 5,334,266, 5,458,700, 5,431,751, 5,053,085 30 and its derived divisional patents (U.S. Pat. Nos. 5,240,517, 5,320,688, and 5,368,658) utilize RSP to achieve either amorphous, a mixture of amorphous and crystalline, or crystalline microstructures, without the use of an L<sub>1</sub><sub>2</sub> strengthening dispersion. The aluminum alloys of U.S. Pat. No. 4,889,582 achieve the high temperature strength with RSP process and age

hardening to produce  $\text{Al}_3\text{Fe}$  (which has non-cubic  $\text{DO}_3$ , not  $\text{L1}_2$  structure) strengthening precipitates, using different alloying components. U.S. Pat. No. 4,464,199 describes high strength aluminum alloys with 4 to 15% iron processed with RSP to achieve high temperature stability. Use of iron promotes  $\text{Al}_3\text{Fe}$  precipitation.

5 The aluminum alloys of U.S. Pat. No. 4,787,943 are composed of transition metals and rare earth metal components. However, with titanium and iron as the preferred transition metals, and gadolinium and cerium as rare earth elements, the alloy does not comprise a stable  $\text{L1}_2$  strengthening dispersion.

U.S. Pat. No. 6,248,453B1 describes the use of an  $\text{L1}_2$  strengthening dispersion for high 10 strength aluminum alloys utilizing certain alloying components, but does not utilize a transition metal such as nickel and cobalt and an intermediate amorphous state as a means to refine the  $\text{L1}_2$  microstructure.

Finally, the high strength and high temperature aluminum alloys referenced have not found widespread application due to the lack of robust processing to produce those alloys and 15 the deficiency of other alloy properties, such as toughness and ductility.

## SUMMARY OF THE INVENTION

The present invention is directed to a new class of aluminum alloys characterized by formation from an intermediate amorphous state to a final face centered cubic (fcc) aluminum 20 matrix strengthened by a precipitate dispersion in order to achieve high temperature strength with usable ductility. The disclosed aluminum alloys thus are comprised of a fine crystalline structure of primarily aluminum fcc matrix strengthened by a precipitate phase with high cubic symmetry, preferably the  $\text{L1}_2$  phase. An appropriate melt of aluminum with crystalline formers is first processed to achieve an intermediate amorphous state to preserve the crystalline 25 formers. The preferred method to achieve a primarily (above 70% in volume) amorphous state is a rapid solidification process (RSP) from the molten alloy by process techniques such as powder atomization, melt spinning, and spray casting. The RSP process preferably should have a cooling rate of at least about  $10^3^\circ\text{C/sec}$ , preferably at least  $10^4^\circ\text{C/sec}$ . Other methods to achieve amorphous microstructure through a solid-state process, such as mechanical milling, 30 may also be used. The intermediate amorphous alloys are then devitrified to a final primarily (above 70% in volume) fcc/ $\text{L1}_2$  crystalline microstructure with at least about 30% fcc phase in volume and at least about 10%  $\text{L1}_2$  phase in volume. Thermal treatment, thermomechanical

treatment, or a combination of both, are the preferred methods to produce the final crystalline microstructure.

The alloys of the invention demonstrate high strength capability after both short and long exposure to a high temperature at or above 300°C by maintaining a fine and stable 5 microstructure. For example, a short-term strength requirement of 40 – 60 ksi yield strength is expected. For the long-term strength, as characterized by creep strength, it has been demonstrated that optimal creep resistance at 300°C for aluminum alloys is obtained from a dispersion of 25 nm diameter L<sub>1</sub><sub>2</sub> particles [E. A. Marquis, *Microstructural Evolution and Strengthening Mechanisms in Al-Sc and Al-Mg-Sc Alloys*, Ph.D. thesis, Northwestern 10 University, 2002.]. Also, it has been demonstrated [Ohdera, Inoue and Masumoto, in *First International Conference on Processing Materials for Properties*, Ed. Henein and Oki, The Minerals, Metals & Materials Society, p.713, 1993] that aluminum-based alloys devitrified by extrusion from amorphous atomized powder can achieve aluminum fcc grain size of 80 to 170 nm, with intermetallic phase dispersions as fine as 10 to 80 nm in diameter. Such grain or 15 particle sizes may therefore encompass the desired optimal particle diameter for creep strengthening by an L<sub>1</sub><sub>2</sub> phase intermetallic dispersion. Accordingly, a process and an aluminum alloy product combining a sufficient amount of stable L<sub>1</sub><sub>2</sub> refined or converted by devitrification from a glass phase material by appropriate thermal or thermomechanical processing has been discovered. The alloy meets or exceeds high temperature strength 20 requirements with optimized creep resistance as a result of intermetallic, small particle phase dispersions.

The aluminum alloys of this invention are produced by combining aluminum, selected metals, particularly transition metals, rare earth elements, and other optional elements into a melt which is converted to an amorphous or glassy state and then subsequently to a crystalline 25 state by a devitrification process. In the broadest sense about 2 to 15 atomic percent in the aggregate of at least one rare earth selected from the group consisting of erbium (Er), thulium (Tm), ytterbium (Yb) and lutetium (Lu) is provided for the melt. Further about 2 to 7 atomic percent in the aggregate of at least one transition metal selected from the group consisting of nickel (Ni), cobalt (Co), iron (Fe) and copper (Cu) is provided to enhance or facilitate the 30 process step of forming an amorphous or glassy material from the melt. Other materials included, optionally, in the melt comprise additional transition metals, lithium (Li) and magnesium (Mg). For example, of less than about 5% in the aggregate by atomic fraction scandium (Sc), yttrium (Y), titanium (Ti), zirconium (Zr), vanadium (V), chromium (Cr),

manganese (Mn) and lithium (Li) incorporated into the melt or alloy as a modifier. Less than about 7% in the aggregate by atomic fraction magnesium (Mg), zinc (Zn), silver (Ag) and niobium (Nb) may be incorporated into the melt or alloy to accommodate fcc misfit in the crystalline state.

5       The selection of alloying components is based on the function or criteria of (1) good L<sub>1</sub><sub>2</sub> formers, (2) adequate glass formability with RSP — this leads to the selection of components with strong short range ordering effects and slow long-range diffusing kinetics in molten aluminum, and (3) good fcc/L<sub>1</sub><sub>2</sub> coherency with low misfit which lead to a fine microstructure and good coarsening resistance at high temperature.

10      Lanthanide rare earth L<sub>1</sub><sub>2</sub> formers such as erbium (Er), thulium (Tm), ytterbium (Yb), and lutetium (Lu) are slow diffusers and hence enhance glass formability while at the same time allowing L<sub>1</sub><sub>2</sub> formation during devitrification. Scandium (Sc) is a strong L<sub>1</sub><sub>2</sub> former which can be used in conjunction with rare earth components to promote L<sub>1</sub><sub>2</sub> formation and reduce the misfit between fcc and L<sub>1</sub><sub>2</sub> phases. Transition metals such as nickel (Ni), cobalt (Co), iron 15 (Fe), and copper (Cu) can enhance short range ordering in the molten aluminum, leading to the increase of glass transition temperature and promoting glass formability. Magnesium (Mg), as a component with relatively high solubility in crystalline aluminum, can be used in fcc aluminum to further reduce the fcc/L<sub>1</sub><sub>2</sub> misfit.

20      With efficient precipitation strengthening by the L<sub>1</sub><sub>2</sub> dispersion, the alloys of the subject invention achieve a hardness of 490 VHN at significantly higher aging temperature compared to conventional aluminum alloys indicating a usable strength at or beyond 300°C.

25      Thus, it is an object of the invention to provide a new class of high-strength L<sub>1</sub><sub>2</sub> phase strengthened aluminum alloys processed into the amorphous state, preferably with a rapid solidification process, and then subsequently devitrified into a crystalline structure with a thermomechanical or thermal processes.

30      A further object of the invention is to provide aluminum alloys with usable strength at or above 300°C by selecting (1) slow diffusing L<sub>1</sub><sub>2</sub> formers to prevent fast coarsening kinetics at high temperature, (2) appropriate matrix fcc and L<sub>1</sub><sub>2</sub> components to reduce the lattice misfit between them to further slow down the coarsening kinetics and to promote the initial finer L<sub>1</sub><sub>2</sub> morphology.

Another object of the invention is to utilize the cubic crystal structure of L<sub>1</sub><sub>2</sub> phase to promote the interfacial coherency with an aluminum fcc matrix and the inherent ductile

behavior of the L<sub>1</sub><sub>2</sub> crystalline phase itself, due to high number of slip systems in cubic L<sub>1</sub><sub>2</sub> structure, to maintain reasonable overall alloy ductility.

A further object of the invention is to combine alloying components so that the L<sub>1</sub><sub>2</sub> formation is enhanced while the likelihood of forming other undesirable intermetallics is  
5 reduced, in order to further promote alloy ductility of an aluminum alloy.

Another object of the invention is to provide adequate glass formability during the rapid solidification process such as powder atomization or melt spinning to form an amorphous aluminum alloy structure, preserving the L<sub>1</sub><sub>2</sub> formers in the solution before a subsequent devitrification process.

10 A further object of the invention is to utilize the intermediate, highly supersaturated amorphous state, slow L<sub>1</sub><sub>2</sub> coarsening kinetics, and low L<sub>1</sub><sub>2</sub>/fcc misfit to produce a fine microstructure in a strong aluminum alloy.

15 A further object of the invention is to provide aluminum alloys having high strength at high temperatures with appropriate toughness when processed with powder metallurgy techniques through an amorphous state.

Another object of the invention is to combine transition metals and rare earth components to provide sufficient glass formability during a rapid solidification process and a nanoscale L<sub>1</sub><sub>2</sub> dispersion strengthened aluminum alloy fcc crystalline structure after devitrification.

20 These and other objects, advantages and features will be set forth in the detailed description which follows.

#### BRIEF DESCRIPTION OF THE DRAWING

In the detailed description which follows, reference will be made to the drawing  
25 comprised of the following figures:

Figure 1 is a chart setting forth criteria for the design of aluminum alloys of the invention;

30 Figure 2 is a diagram illustrating the viscosity of liquid aluminum nickel alloys calculated at 1700°C in comparison with experimental data wherein the comparison is based upon calculations made using different thermodynamic descriptions of the aluminum nickel liquid (substitutional solution model without associate or associate model with AlNi associate);

Figure 3 is an assessed Al – Yb phase diagram in comparison with experimental data;

**Figure 4a** is a graph of the enthalpy of formation of cobalt and a rare earth, i.e. Er and Yb predicted by quantum mechanical calculation;

**Figure 4b** is a graph of the mixing enthalpy of the materials set forth in Figure 4a;

**Figure 4c** is a graph of the molar volume of the materials set forth in Figure 4a;

5       **Figure 5** is an X-ray diffractogram of the alloy of Example 1A as melt-spun with positions of fcc pure aluminum reflections indicating a fully amorphous state;

**Figure 6** is an isochronic DSC thermogram of the alloy of Example 1A indicating devitrification steps upon heating;

10      **Figure 7** is an X-ray diffractogram of the alloy of Example 1A after devitrification at 425°C for 22 hours, with positions of reflections of pure fcc Al, Al<sub>3</sub>Yb, and Al<sub>3</sub>Er phases, indicating the desired phases: FCC+L<sub>1</sub><sub>2</sub>;

**Figure 8** is a SIMS elemental map of devitrified alloy of Example 1A indicating Er and Yb partition together to L<sub>1</sub><sub>2</sub> phase, and Ni and Co which do not partition to L<sub>1</sub><sub>2</sub>;

15      **Figure 9** is a scanning electron microscope (SEM) secondary electron image of devitrified alloy of Example 1A indicating submicron morphology;

**Figure 10** is a 3DAP microscopy reconstruction of Al<sub>3</sub>(Yb,Er)L<sub>1</sub><sub>2</sub> phase of the alloy of Example 1A after devitrification at 424°C for 19 hours;

**Figure 11** is an X-ray diffractogram of the alloy of Example 2A as melt-spun, with positions of fcc pure Al and L<sub>1</sub><sub>2</sub> reflections, indicating fully amorphous status;

20      **Figure 12** is an X-ray diffractogram of the alloy of Example 2A after devitrification at 425°C for 19 hours, with positions of reflections of pure fcc Al, Al<sub>3</sub>Yb, and Al<sub>3</sub>Er phases, indicating the desired phases: FCC+L<sub>1</sub><sub>2</sub>;

**Figure 13** is an X-ray diffractogram of the alloy of Example 2B as melt-spun, with positions of fcc pure Al and L<sub>1</sub><sub>2</sub> reflections, indicating fully amorphous status;

25      **Figure 14** is an X-ray diffractogram of the alloy of Example 2B after devitrification at 425°C for 19 hours, with positions of reflections of pure fcc Al, Al<sub>3</sub>Yb, and Al<sub>3</sub>Er phases, indicating the desired phases: FCC+L<sub>1</sub><sub>2</sub> wherein the lattice parameter of L<sub>1</sub><sub>2</sub> is reduced by Sc;

**Figure 15** is an X-ray diffractogram of the alloy of Example 2C as melt-spun, with positions of fcc pure Al and L<sub>1</sub><sub>2</sub> reflections, indicating almost fully amorphous status;

30      **Figure 16** is an X-ray diffractogram of the alloy of Example 2C after devitrification at 425°C for 19 hours, with positions of reflections of pure fcc Al, Al<sub>3</sub>Yb, and Al<sub>3</sub>Er phases, indicating the desired phases: FCC+L<sub>1</sub><sub>2</sub> wherein the lattice parameter of L<sub>1</sub><sub>2</sub> is expanded by Mg;

Figure 17 is an X-ray diffractogram of the alloy of Example 2D as melt-spun, with positions of fcc pure Al and L<sub>1</sub><sub>2</sub> reflections, indicating fully amorphous status;

Figure 18 is an X-ray diffractogram of the alloy of Example 2D after devitrification at 425°C for 19 hours, with positions of reflections of pure fcc Al, Al<sub>3</sub>Yb, and Al<sub>3</sub>Er phases, 5 indicating the desired phases: FCC+L<sub>1</sub><sub>2</sub> and wherein the lattice parameter of fcc is expanded by Mg and that of L<sub>1</sub><sub>2</sub> is reduced by Sc;

Figure 19 is an X-ray diffractogram of the alloy of Example 1C as melt-spun, with positions of fcc pure Al and L<sub>1</sub><sub>2</sub> reflections, indicating non-amorphous status due to the absence of transition metal elements;

10 Figure 20 is an X-ray diffractogram of the alloy of Example 1C after devitrification at 425°C for 19 hours, with positions of reflections of pure fcc Al, Al<sub>3</sub>Yb, and Al<sub>3</sub>Er phases, indicating elements ER, Yb, and Sc can be completely intersoluble to form L<sub>1</sub><sub>2</sub> and wherein the lattice parameter of L<sub>1</sub><sub>2</sub> is reduced by Sc; and

15 Figure 21 is a scanning electron microscope (SEM) secondary electron image of a devitrified alloy of Example 1C showing coarse microstructure in comparison with the alloy of Example 1A (see Figure 9) and indicating that passing through the glassy state can reduce the particle size of precipitation.

## DESCRIPTION OF THE PREFERRED EMBODIMENTS

20

### General Summary

In general, the subject matter of the invention comprises a product (alloy) as well as a product by process and a process all directed to an aluminum alloy in crystalline form having higher or greater strength particularly at elevated temperatures, i.e. greater than at least 250°C 25 and preferably 300°C. The alloy of aluminum is made by compounding a mixture of aluminum with selected rare earths (RE) and transition metals (TM) in an amorphous or glassy state followed by devitrification of the material from the amorphous state to a mixed crystalline state comprised of face centered cubic (fcc) and L<sub>1</sub><sub>2</sub>. (an ordered fcc) state wherein the ratios of the crystalline states is within certain preferred ranges. Preferably the resultant alloy has at least 30 about 30% by volume fcc phase and at least about 10% by volume L<sub>1</sub><sub>2</sub> phase material with limited residual amorphous or other phase material such as quasi-crystalline or ordered amorphous material.

The choice of starting materials may be varied as may the compounding processes, the glass phase formation processes and the devitrification processes. Importantly, when in the amorphous state, there may be some crystalline material contained therein, but preferably no more than about 30% by volume in the fcc phase. Thus forming the mixture in the amorphous 5 or glassy intermediate state and subsequent devitrification constitute very important aspects of the invention.

The alloy materials, in addition to aluminum (Al), include one or more lanthanide rare 10 earths (RE) selected or taken from the group of erbium (Er), thulium (Tm), ytterbium (Yb), and lutetium (Lu), and one or more transition metals (TM) taken or selected from the group of copper (Cu), nickel (Ni), cobalt (Co), titanium (Ti), iron (Fe), yttrium (Y), and scandium (Sc). Further additives in minor amounts (less than about five percent (5%) atomic percent) may be included such as magnesium (Mg). The range of various constituents is also important to 15 maintain processability and preserve transition characteristics. Thus RE materials are utilized in the range of about 2 to 15 atomic percent (%), and the TM and metals are utilized in the aggregate range of about 2 to 12 atomic percent (%). Transition metals (TM) in the range of about 2 to 7 atomic percent are utilized.

The processes for mixing or forming the starting materials in the amorphous state are not necessarily limiting. Thus, it is contemplated that solid state processing, liquid or melt 20 processing as well as gas phase processing may be utilized, though liquid phase processing is preferred based upon examples as reported hereinafter. The completeness of the amorphous state is preferably at least about 70% by volume and preferably greater. The amorphous state is a solid noncrystalline state or phase prior to devitrification.

#### Development Technique

Devitrification is accomplished via heat treating (thermal) processing, thermo- 25 mechanical processing as well as other processing techniques.

Precipitation strengthened aluminum (Al) alloys are difficult to develop for high strength due to limited solubility of alloying elements in fcc Al. The low solubility of strengthening dispersion elements prevents the solid-state solution treatment of the alloy to bring strengthening components into solution. It is hence not necessarily possible to allow subsequent aging to 30 produce a desirable strengthening microstructure. Al alloys with high fractions of precipitate that cannot be completely solution treated therefore have very coarse particle sizes that tend to limit strength, corrosion resistance and toughness. In contrast, the Al alloys exhibit composition

and microstructure characteristics that enable the Al alloys of invention to exhibit high strength, good ductility, and high temperature stability at or above 300°C.

By selecting an appropriate composition, processing techniques achieve a fully amorphous state after rapid cooling of the Al alloy composition. Furthermore, with careful 5 selection of composition this glass can then be thermal or thermo-mechanically processed such that the glass devitrifies into a fully crystalline fcc matrix with intermetallic precipitates having a very fine size scale. By passing through the glass state, the equilibrium solidification that would produce coarse precipitates is avoided. Furthermore, the selection of the precipitate phase as L1<sub>2</sub> allows the precipitate to be coherent or semi-coherent with the fcc matrix 10 improving the ductility of the alloy and the resistance of the precipitate to coarsen during subsequent thermal treatment.

Figure 1 is a system flow-block diagram that illustrates the processing/structure/properties relationships for alloys of the invention. The two-stage nature of the chart emphasizes the generally equal importance of processability and final performance in 15 the design of alloys of the invention. The processability is governed by the glass forming ability; strength, ductility, and thermal stability are the required end properties.

Certain transition metals (TM) (such as Fe, Co, Ni, Cu) promote short-range ordering in liquid Al, which leads to low partial molar volume, low thermal expansion, and high viscosity that are beneficial to glass forming ability. Rare-earth (RE) elements (such as Ce, Gd, Yb, Er) 20 with large atomic size exhibit low diffusivity in Al and thus retard crystal nucleation. Therefore, Al-TM-RE comprise a class of glass forming system for Al alloys.

Viscosity of liquid metals also has a strong effect on the glass forming ability. Liquid phase metals having high viscosity usually becomes glass through fast cooling. A binary model to describe liquid's viscosity is set forth in Equation (1):

$$25 \quad \eta = -\frac{x_1 x_2}{RT} (x_1 \eta_1^\circ + x_2 \eta_2^\circ) \frac{\partial^2 G_m}{\partial x_2^2} \quad (1)$$

The variables  $x_i$  and  $G_m$  represent the mole fraction and molar Gibbs energy of the liquid, respectively. The viscosity of pure components,  $\eta_i^\circ$ , is available from literature, see Table 1. From the above equation it is seen that the composition dependence of the viscosity is fully determined by the thermodynamic properties of the liquid.

Table 1. Viscosity and related data for pure liquid metals  
 [L. Battezzati and A.L. Greer, "The Viscosity of Liquid Metals and Alloys," *Acta metall.* 37, 1791 (1989).]

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Element	T <sub>m</sub> (K)	V <sub>m</sub> (10 <sup>-6</sup> m <sup>3</sup> /mol)	η(T <sub>m</sub> ) (10 <sup>-3</sup> Pa s)	C <sub>A</sub> [10 <sup>-7</sup> (J/K mol <sup>1/3</sup> ) <sup>1/2</sup> ]	E (kJ/mol)	B (E/RT <sub>m</sub> )	η <sub>0</sub> (10 <sup>-3</sup> Pa s)
Al	933	11.3	1.30	1.30	16.5	2.13	0.149
Co	1765	7.59	4.18	1.58	44.4	3.03	0.255
Cr	2133	8.28	5.7	~2.2	~185	~10.43	1.7x10 <sup>-4</sup>
Cu	1356	7.94	4.0	1.72	30.5	2.71	0.301
Fe	1809	7.96	5.5	2.18	41.4	2.75	0.370
Mn	1517	9.58	5	2.5	20-46.5	1.6-3.7	0.12-1.02
Ni	1728	7.43	4.90	1.85	50.2	3.49	0.166
V	2175	8.9	2.4	0.98	~73	~4.04	~0.042
Ti	1958	11.6	2.2	1.18	~68	~4.18	~0.034
Zr	2125	~15.7	3.5	1.58	~88	~4.98	~0.024

T<sub>m</sub>: melting temperatureV<sub>m</sub>: molar volume at T<sub>m</sub>η(T<sub>m</sub>): viscosity at T<sub>m</sub>C<sub>A</sub>: constant

E: activation energy for viscous flow

B: B = E/RT<sub>m</sub>η<sub>0</sub>: pre-exponential viscosity

10       Equation (1) has been validated for several binary systems where both experimental viscosity data and thermodynamic descriptions are available. An example is the Al-Ni system where an intermediate compound AlNi melts congruently at about 1650°C. Figure 2 shows the calculated viscosity in Al-Ni liquid at 1700°C as a function of Ni content, in comparison with experimental data. It is seen that the calculated viscosity shows a similar trend as the 15 experimental data, but the latter indicates a stronger increase of the viscosity around 50 at.% Ni. The strong increase in viscosity is believed to be caused by molecular associates of Al-Ni in the liquid. In order to illustrate the effect of the AlNi associates on the viscosity, the Al-Ni system was thermodynamically reassessed with an associate model for the liquid free energy. The viscosity of Al-Ni liquid calculated using the AlNi associate model is also plotted in Figure 20 2 for comparison. The calculation shows a significant increase in the viscosity due to the formation of liquid AlNi associates.

This viscosity model has further been extended to multi-component systems, which provides guidance of glass forming ability for the design of alloys of the invention.

25       A model to predict glass transition temperature, T<sub>g</sub>, has been developed for the Al-TM-RE system. The model is based on the parabolic relationship between T<sub>g</sub> and room temperature glass shear modulus, G, and it assumes that G is linear in atomic fraction, X, of each element according to equation (2):

$$T_g = \left( 14100 \sum_i K_i X_i \right)^{0.49} \quad (2)$$

*i = Al, Mg, Ni, Co, Er, Yb, Sc, Y, etc.*

The coefficient for each element,  $K_i$ , has been calibrated to experimental DSC data of melt-spun amorphous ribbons. These coefficients quantitatively represent the relative effectiveness of the aforementioned molecular associates in promoting glass forming ability by varying  $T_g$ . This model also provides guidance of glass forming ability for the design of alloys of the invention. The prediction error of equation (2) compared to available experimental data is about 9 K with a standard deviation of 10 K, which is within the experimental precision.

The desired performance for applications (e.g. aircraft structural components, high temperature turbine components, etc.) determines a set of alloy properties required, including strength, ductility and thermal stability resistance up to 300°C. The desired properties can be achieved through a fine, stable, multicomponent  $L1_2$  phase dispersion in the Al matrix after devitrification of the glassy Al, as long as a sufficient volume fraction of  $L1_2$  and low misfit with the Al matrix and thus low coarsening rate can be obtained.

The elements Er, Lu, Tm and Yb are known as the only stable rare earth (RE)  $L1_2$  formers. The melting point and cost of these rare earth elements are presented in Table 2. Some properties of corresponding trialuminides are also given in Table 2, including lattice parameter, density, and  $L1_2$  decomposition or congruent melting temperature. Among the four RE elements, according to Table 2, Yb has the smallest lattice parameter and relatively lower cost. Er has the lowest cost. Both Yb and Er have relatively lower melting point. Usually, the lower the melting temperature, the higher the glass forming ability. As a consequence, Alloys of the invention utilize Yb and Er as preferred  $L1_2$  formers rather than Tm and Lu.

Table 2. Stable  $L1_2$  formers and their properties.

Element	Melting Point	Cost <sup>a</sup>	$L1_2$	Lattice Parameter	Density	$L1_2$ Max. Temp.
Er	1529°C	\$725/kg	Al <sub>3</sub> Er	4.226 Å	5.507 g/cm <sup>3</sup>	1070°C
Lu	1663°C	\$7500/kg	Al <sub>3</sub> Lu	4.186 Å	5.783 g/cm <sup>3</sup>	1200°C
Tm	1545°C	\$6500/kg	Al <sub>3</sub> Tm	4.203 Å	5.591 g/cm <sup>3</sup>	
Yb	819°C	\$1600/kg	Al <sub>3</sub> Yb	4.202 Å	5.682 g/cm <sup>3</sup>	980°C
Sc	1541°C	\$18,000/kg	Al <sub>3</sub> Sc	4.101 Å	3.03 g/cm <sup>3</sup>	1320°C

<sup>a</sup> For 1-5 kg cast metal ingots from 99.9%-grade oxides

Sc is the only transition metal (TM) element that can form a stable  $L1_2$  with Al. In comparison with RE  $L1_2$  formers, Sc can form  $L1_2$  with a smaller lattice parameter of 4.101 Å,

thus leading to smaller misfit between L1<sub>2</sub> and the Al matrix. However, Sc is by far the most expensive of the elements considered, as seen in Table 2. The alloy of the invention therefore seeks to limit the use of Sc as much as possible. Efforts have been made to search for other transition metals to substitute for Sc. A preliminary requirement for such substitution is 5 solubility. Experiments have shown that Ti has a substantial solubility in Al<sub>3</sub>Sc, up to 12.5 at.%, i.e. Al<sub>0.75</sub>Sc<sub>0.125</sub>Ti<sub>0.125</sub>. In addition, Ti has the lowest diffusion coefficient in solid Al among the transition metals. For instance, the diffusivity in Al at 300°C is 9.0×10<sup>-20</sup> m<sup>2</sup>/s for Sc, 6.3×10<sup>-24</sup> m<sup>2</sup>/s for Zr, and 2.7×10<sup>-25</sup> m<sup>2</sup>/s for Ti. Adding Ti to Al<sub>3</sub>Sc will thus reduce the coarsening rate of L1<sub>2</sub> precipitates. Moreover, addition of Ti decreases the lattice parameter of 10 Al<sub>3</sub>(Sc,Ti) and hence minimizes the lattice misfit with Al. The lattice parameters for seven hypothetical L1<sub>2</sub> compounds with transition metals are calculated at zero degree Kelvin (0 K) using the first-principle quantum mechanical approach. Results are shown in Table 3. All seven hypothetical L1<sub>2</sub> compounds show significant smaller lattice parameters than those stable L1<sub>2</sub> phases. Thus, alloys of the invention incorporate Yb, Er and Sc as base L1<sub>2</sub> formers but are not 15 limited to these elements. TMs such as Ti, V, Zr, etc, which will result in low misfit and thus lower interfacial energy to yield a low driving force for particle coarsening are considered useful.

Table 3 Lattice Parameters of hypothetical L1<sub>2</sub> TM Compounds through First Principle Calculations at 0 K.

20

TM <sub>s</sub> Hypothetic L1 <sub>2</sub>	Lattice Parameter, Å
Al <sub>3</sub> Ti	3.967
Al <sub>3</sub> Zr	4.085
Al <sub>3</sub> V	4.045
Al <sub>3</sub> Fe	3.8005
Al <sub>3</sub> Ni	3.8469
Al <sub>3</sub> Co	3.7946
Al <sub>3</sub> Cu	3.9363

25 It should be pointed out that the compound Al<sub>3</sub>Ti is metastable as the L1<sub>2</sub> phase butstable as D0<sub>22</sub>. It can be transformed to the L1<sub>2</sub> structure if Al is substituted partially by Cr,

Mn, Fe, Co or Ni.

For a multicomponent L1<sub>2</sub> phase, the lattice parameter  $\alpha$  can be calculated based on a linear model similar to Vegard's law. As an example, we have the following equation for L1<sub>2</sub> with the formula Al<sub>3</sub>(Er<sub>x</sub>Yb<sub>y</sub>Sc<sub>z</sub>) is utilized:

$$\alpha[\text{Al}_3(\text{Er}_x\text{Yb}_y\text{Sc}_z)] = x * \alpha(\text{Al}_3\text{Er}) + y * \alpha(\text{Al}_3\text{Yb}) + z * \alpha(\text{Al}_3\text{Sc}) \quad (3)$$

Assuming L<sub>1</sub><sub>2</sub> has a linear thermal expansion, its lattice parameter at elevated temperature can be calculated as follows (T in °C):

$$a(L_1_2, T) = a(L_1_2, RT) \times [1 + (T - 25) \times \alpha(L_1_2)] \quad (4)$$

where  $\alpha(L_1_2)$  represents the thermal expansion estimated according to the following reciprocal

5 relation:

$$\frac{\alpha(M)}{\alpha(Al_3M)} = \frac{T_m(Al_3M)}{T_m(M)} \quad (5)$$

where M represents the corresponding pure metal in Al<sub>3</sub>M and  $T_m$  is the melting point (°K).

Table 4 summarizes melting point, thermal expansion of relevant elements in alloys of the invention and the estimated thermal expansion as well as lattice parameter at 25°C and 300°C

10 for stable L<sub>1</sub><sub>2</sub> or meta-stable L<sub>1</sub><sub>2</sub>.

Table 4. Estimation of Thermal Expansion Coefficients and Lattice Parameters.

Element	Melting Point, °K	Thermal Expansion, 1/°K, 20~100°C	Thermal Expansion, 1/°K, 20~300°C	L <sub>1</sub> <sub>2</sub> Alloys	Melting Point, °K	Thermal Expansion, 20~300°C	Lattice Parameter, Å, 25°C	Lattice Parameter, Å, 300°C
Ni	1728	13.3×10 <sup>-6</sup>	14.4×10 <sup>-6</sup>	Al <sub>3</sub> Ni	913	27.254×10 <sup>-6</sup>	3.8639	3.8929
Co	1768	12.5×10 <sup>-6</sup>	13.6×10 <sup>-6</sup>	Al <sub>3</sub> Co	1408	17.077×10 <sup>-6</sup>	3.8114	3.8293
Yb	1092	25×10 <sup>-6</sup>	25×10 <sup>-6</sup>	Al <sub>3</sub> Yb	1253	21.788×10 <sup>-6</sup>	4.202	4.2272
Er	1802	9.2×10 <sup>-6</sup>	9.2×10 <sup>-6</sup>	Al <sub>3</sub> Er	1343	12.344×10 <sup>-6</sup>	4.226	4.2403
Sc	1814	12×10 <sup>-6</sup>	12×10 <sup>-6</sup>	Al <sub>3</sub> Sc	1593	13.665	4.101	4.1164
Ti	1943	8.9×10 <sup>-6</sup>	9.2×10 <sup>-6</sup>	Al <sub>3</sub> Ti	1623	11.014	3.967	3.979
Y	1795	10.8×10 <sup>-6</sup>	10.8×10 <sup>-6</sup>	Al <sub>3</sub> Y	1253	15.472	4.299	4.3173
Al			25.3×10 <sup>-6</sup>				4.0497	4.0779

In the practice of the invention, decrease of the misfit by expanding the lattice parameter of the Al matrix is a useful technique. There are few elements that have significant solubility in 15 Al. Mg is the only one that can substantially increase the lattice spacing of Al. According to literature information, in dilute AlMg alloys, every one atom percent of Mg will expand the Al lattice by 0.0045 Å. Further, every one atom percent of Mg will increase the linear coefficient of thermal expansion of Al by 1.179×10<sup>-7</sup>/°C.

Thermodynamic modeling and assessments have been performed for Al-TM-RE systems with relevant elements incorporated in alloys of the invention. The assessed Al-Yb binary phase diagram is shown in Figure 3. The L<sub>1</sub><sub>2</sub> Al<sub>3</sub>Yb is the strengthening phase that is used in alloys of the invention. According to Figure 3, Al<sub>3</sub>Yb is not solution treatable. The only way to avoid coarse precipitation from Al liquid is using fast cooling. It is possible to get fine

crystalline precipitates directly by rapid solidification processing (RSP) from the melt. However, much finer precipitates can be generated through devitrification of amorphous state due to the larger driving force. In addition, the precipitation process can be controlled through devitrification.

5 Quantum mechanical calculations have been performed in Al<sub>3</sub>Er-Al<sub>3</sub>Co and Al<sub>3</sub>Yb-Al<sub>3</sub>Co systems. The predicted enthalpy of formation, mixing enthalpy and molar volume in the corresponding systems are plotted in Figure 4. According to Figure 4, there is a strong repulsive Co-RE interaction indicating negligible Co solubility in L1<sub>2</sub> Al<sub>3</sub>RE. This sets a challenge in producing good glass forming alloys through conventional technology in the same  
10 composition intended to precipitate a significant amount of L1<sub>2</sub> precipitation through devitrification.

#### Experimental Results

The present invention alloys, through computational design of multi-component Al-TM-RE systems incorporate, desired processing properties – glass forming ability and the  
15 desired microstructure – a fine dispersion of L1<sub>2</sub> after devitrification in the Al matrix. Alloys of the invention are summarized in Table 5 for their composition, status in melt-spun and devitrification conditions, as well as the lattice parameter of the Al fcc and the L1<sub>2</sub> phases and the misfit between L1<sub>2</sub> and Al matrix.

Table 5 Summary of Composition and Microstructure of Al Alloys

No.	Alloy Composition at.%	As-spun status	Main phases after devitrification	Lattice Parameter		Misfit %
				L1 <sub>2</sub> , a(Å)	fcc, a(Å)	
1A	Al-2.5Ni-2.5Co-3.5Yb-3.5Er	fully glass	fcc, L1 <sub>2</sub> + minor x phase	4.215	4.063	3.735
1B	Al-2.75Ni-2.75Co-2.17Yb-2.17Er-2.17Y	fully glass	fcc, L1 <sub>2</sub> +Al <sub>3</sub> Y+ minor x phase	4.217	4.050	4.123
1C	Al-6Yb-6Er	fcc, L1 <sub>2</sub> + some glass	fcc, L1 <sub>2</sub>	4.208	4.057	3.726
1D	Al-4Yb-4Er-4Y	fcc, L1 <sub>2</sub> + some glass	fcc, L1 <sub>2</sub> + Al <sub>3</sub> Y	4.229	4.059	4.192
1E	Al-4.5Yb-4.5Er-3Sc	fcc, L1 <sub>2</sub> + some glass	fcc, L1 <sub>2</sub>	4.183	4.060	3.032
1F	Al-12Er-3Mg	fcc, L1 <sub>2</sub> + some glass	fcc, L1 <sub>2</sub>	4.216	4.059	3.865
1G	Al-12Er-6Mg	fcc, L1 <sub>2</sub> + some glass	fcc, L1 <sub>2</sub>	4.217	4.065	3.739
1H	Al-12Yb-3Mg	fcc + some glass	fcc, L1 <sub>2</sub>	4.216	4.076	3.435
1I	Al-12Yb-6Mg	fcc, L1 <sub>2</sub> + some glass	fcc, L1 <sub>2</sub>	4.191	4.061	3.201
2A	Al-2Ni-2Co-4Yb-4Er	almost fully glass+minor fcc	fcc, L1 <sub>2</sub> + minor x phase	4.201	4.049	3.754
2B	Al-2.5Ni-2.5Co-2Yb-2Er-3Sc	almost fully glass+minor fcc	fcc, L1 <sub>2</sub> + minor x phase	4.150	4.048	2.520
2C	Al-3Mg-2.5Ni-2.5Co-4.5Yb-2.5Er	mostly glass + minor fcc	fcc, L1 <sub>2</sub> + minor x phase	4.197	4.060	3.374
2D	Al-3Mg-2.5Ni-2.5Co-2.5Yb-1.5Er-3Sc	fully glass	fcc, L1 <sub>2</sub> + minor x phase	4.155	4.057	2.416

According to Table 5, alloys with TM elements (e.g. Ni and Co) are fully glass or almost fully glass in the as melt-spun conditions. These alloys include alloy 1A, 1B, 2A, 2B, 5 2C and 2D. Alloys without Ni and Co primarily show FCC Al and L1<sub>2</sub> crystalline phases although they are partially amorphous in the as melt-spun conditions. These alloys include alloy 1C, 1D, 1E, 1F, 1G, 1H and 1I. The results confirm the prediction that TM elements such as Ni and Co are critical elements to promote the glass forming ability in the alloys of the invention. Through devitrification, all alloys precipitate mainly L1<sub>2</sub> in the Al matrix. It should be emphasized that the particle size of alloys passing through a fully or almost fully glassy state is much finer than that of alloys without passing through the glassy state or only passing through a partially glassy state with L1<sub>2</sub> already present in the as-spun condition.

Following is a detailed demonstration of characteristics of alloys in Table 5.

**Example 1**

Alloy 1A in Table 5 was fabricated into both powders using gas atomization and ribbons using melt spinning technology (wheel speed 55 m/s). In the as melt-spun 5 condition, Alloy 1A was initially characterized by X-ray diffraction (XRD) and differential scanning calorimetry (DSC). The diffractometer trace (Cu-K $\alpha$  radiation ) of as-spun Alloy 1A in Figure 5 shows a broad peak characteristic of glass, and the ribbons show very good ductility, also characteristic of the glassy state. A DSC trace run at 40°C/min. shown in Figure 6 indicates three stages of glass devitrification at temperatures near 200°C, 315°C and 370°C.

10 DSC thermograms indicated that for heating at a constant rate of 10-40°C/min, the dominant crystallization reactions occur below about 400°C for all alloys under investigation. Based on this information, a temperature of 425°C was selected for the devitrification anneal. Specifically, alloy samples were encapsulated in quartz tubes under a protective atmosphere of approximately 500 mbar Argon with 99.999% purity. The 15 encapsulated specimens were kept at 425°C for periods of 19 or 22 hrs, respectively, to ensure completion of the crystallization reaction and grain growth to a size that facilitates experimental characterization.

Figure 7 shows an X-ray diffractogram of Alloy 1A after devitrification at 425°C for 22 hours. At this point transformations I-III are fully completed and the ribbons are 20 fully devitrified in the sense that no significant amount of amorphous material can be detected by XRD and TEM. The peak positions of (111), (200), and (220) reflections calculated for FCC pure Al (lattice parameter 4.0496 Å) are indicated in red. Within the accuracy of this experiment (that is 0.01° at  $2\theta = 38.505^\circ$  (Al (111))), the measured peak positions are identical to pure Al, indicating the equilibrium solubility of Ni, Co, Yb, and Er in Al is very small. Also 25 marked in red are the positions of the 'forbidden' reflections (110), (210), (211), and (221) for FCC Al, which are superlattice reflections of the L1<sub>2</sub> structure. The peak positions for L1<sub>2</sub> Al<sub>3</sub>Yb (lattice parameter 4.202 Å marked in green, and that for L1<sub>2</sub> Al<sub>3</sub>Er (lattice parameter 4.215 Å) are marked in brown. It can be seen that the measured peak positions are perfectly matched by L1<sub>2</sub> structure reflections. Therefore, Alloy 1A mainly consists of FCC Al and L1<sub>2</sub> 30 phases as devitrified microstructure. It is also noticed that there is a minor third phase x, which crystalline structure is to be identified.

The FCC Al+L<sub>1</sub><sub>2</sub> microstructure of Alloy 1A was investigated by the secondary ion mass spectrometry (SIMS) technology. Figure 8 indicates that Yb and Er partition together into L<sub>1</sub><sub>2</sub> precipitates, while Ni and Co have limited partitions.

The morphology of FCC+L<sub>1</sub><sub>2</sub> microstructure of Alloy 1A is shown in Figure 9.

5 This SEM secondary electron image indicates that the majority of L<sub>1</sub><sub>2</sub> precipitates are less than 1 micron after devitrification at 425°C, 22 hours. Based upon previously discussed experience in devitrified Al alloys, this indicated the potential to obtain nanoscale L<sub>1</sub><sub>2</sub> precipitates with thermomechanical processing.

10 The three dimension atom probe microscopy (3DAPM) analysis confirms that L<sub>1</sub><sub>2</sub> exists in Alloy 1A with formula Al<sub>0.72</sub>(Yb,Er)<sub>0.28</sub>. Yb and Er have about the same site fraction. Ni and Co are negligible in L<sub>1</sub><sub>2</sub> phase. See Figure 10.

15 Alloy 1A was also studied through microhardness testing in ribbon conditions after thermal cycles required for powder consolidation and extrusion. Results are presented in Table 6. Through devitrification at a temperature near 300°C, Alloy 1A has demonstrated a hardness of 490 VHN, which is equivalent to the room temperature tensile strength exceeding 1000 MPa according to Inoue's work in similar alloy systems. (Akio Inoue, "Amorphous, nanoquasicrystalline and nanocrystalline alloys in Al-based systems", Progress in Materials Science 43 (1998) 365-520). This result indicates high efficient strengthening of L<sub>1</sub><sub>2</sub>.

20 Table 6 Microhardness testing of Alloy 1A

Condition	Hardness (VHN)
As-spun	369
177°C 1hr	276
177°C 10hrs	310
177°C 20 hrs	335
177°C 20hrs then 288°C 2hrs.	490
177°C 20hrs then 316°C 2hrs.	340
177°C 20hrs then 400°C 2hrs.	275

### Example 2

Alloy 2A in Table 5 was fabricated into ribbons using melt spinning technology (wheel speed 55 m/s). In the as melt-spun condition, Alloy 2A was characterized by X-ray diffraction. The diffractometer trace (Cu-K $\alpha$  radiation ) of as-spun Alloy 2A in Figure 11 shows a broad

peak characteristic of glass, and the ribbons show reasonable good ductility, also characteristic of the glassy state.

Figure 12 shows an X-ray diffractogram of Alloy 2A after devitrification at 425°C for 19 hours. At this point the ribbon is fully devitrified in the sense that no significant amount of amorphous material can be detected by XRD. The peak positions of (111), (200), and (220) reflections calculated for FCC pure Al (lattice parameter 0.40496 Å) are indicated in red. The measured peak positions are identical to pure Al, indicating the equilibrium solubility of Ni, Co, Yb, and Er in Al is very small. Also marked in red are the positions of the 'forbidden' reflections (110), (210), (211), and (221) for FCC Al, which are superlattice reflections of the L<sub>1</sub><sub>2</sub> structure. The peak positions for L<sub>1</sub><sub>2</sub> Al<sub>3</sub>Yb (lattice parameter 4.202 Å) are marked in green, and that for L<sub>1</sub><sub>2</sub> Al<sub>3</sub>Er (lattice parameter 4.215 Å) are marked in brown. It can be seen that the measured peak positions are perfectly matched by L<sub>1</sub><sub>2</sub> structure reflections. Therefore, Alloy 2A mainly consists of FCC Al and L<sub>1</sub><sub>2</sub> phases as devitrified microstructure. It is noticed that the third phase *x*, in Example 2A, is less than in Example 1A, reflecting the desire to reduce the amount of *x* phase by reducing Ni and Co in Alloy 2A while retaining reasonable glass forming ability.

### Example 3

Alloy 2B in Table 5 was fabricated into ribbons using melt spinning technology (wheel speed 55 m/s). In the as melt-spun condition, Alloy 2B was characterized by X-ray diffraction. The diffractometer trace (Cu-K $\alpha$  radiation) of as-spun Alloy 2B in Figure 13 shows a broad peak characteristic of glass, and the ribbons show reasonable good ductility, also characteristic of the glassy state.

Figure 14 shows an X-ray diffractogram of Alloy 2B after devitrification at 425°C for 19 hours. At this point the ribbon is fully devitrified in the sense that no significant amount of amorphous material can be detected by X-ray diffraction (XRD). The peak positions of (111), (200), and (220) reflections calculated for FCC pure Al (lattice parameter 4.0496 Å) are indicated in red. The measured peak positions are identical to pure Al, indicating the equilibrium solubility of Ni, Co, Yb, Er and Sc in Al is very small. Also marked in red are the positions of the 'forbidden' reflections (110), (210), (211), and (221) for FCC Al, which are superlattice reflections of the L<sub>1</sub><sub>2</sub> structure. It can be seen that the measured peak positions are matched by L<sub>1</sub><sub>2</sub> structure reflections with lattice parameter 4.150 Å. It demonstrates that Yb, Er and Sc can be completely intersoluble to form L<sub>1</sub><sub>2</sub>, and the lattice parameter has been

significantly reduced by adding Sc, which is another means of reducing misfit to keep the L1<sub>2</sub> precipitates coarsening resistant. The misfit between FCC Al and L1<sub>2</sub> is 2.52% for Alloy 2B.

#### Example 4

5       Alloy 2C in Table 5 was fabricated into ribbons using melt spinning technology (wheel speed 55 m/s). In the as melt-spun condition, Alloy 2C was characterized by X-ray diffraction. The diffractometer trace (Cu-K $\alpha$  radiation) of as-spun Alloy 2C in Figure 15 shows a broad peak characteristic of glass, and the ribbons show reasonable good ductility, also characteristic of the glassy state.

10      Figure 16 shows an X-ray diffractogram of Alloy 2C after devitrification at 425°C for 19 hours. At this point the ribbon is fully devitrified in the sense that no significant amount of amorphous material can be detected by XRD. The peak positions of (111), (200), and (220) reflections calculated for FCC pure Al (lattice parameter 4.0496 Å) are indicated in red. The measured peak positions indicate that the lattice space of Al has been dilated to 4.060 Å, 15 reflecting another feature of the invention: adding Mg to expand the lattice space of Al and thus decrease the misfit between Al and L1<sub>2</sub>. Also marked in red are the positions of the ‘forbidden’ reflections (110), (210), (211), and (221) for FCC Al, which are superlattice reflections of the L1<sub>2</sub> structure. It can be seen that the measured peak positions are perfectly matched by L1<sub>2</sub> structure reflections. Therefore, Alloy 2C mainly consists of FCC Al and L1<sub>2</sub> phases as 20 devitrified microstructure.

#### Example 5

Alloy 2D in Table 5 was fabricated into ribbons using melt spinning technology (wheel speed 55 m/s). In the as melt-spun condition, Alloy 2D was characterized by X-ray diffraction. 25 The diffractometer trace (Cu-K $\alpha$  radiation) of as-spun Alloy 2D in Figure 17 shows a broad peak characteristic of glass, and the ribbons show very good ductility, also characteristic of the glassy state.

Figure 18 shows an X-ray diffractogram of Alloy 2D after devitrification at 425°C for 19 hours. At this point the ribbon is fully devitrified in the sense that no significant amount of 30 amorphous material can be detected by XRD. The peak positions of (111), (200), and (220) reflections calculated for FCC pure Al (lattice parameter 4.0496 Å) are indicated in red. The measured peak positions indicate that the lattice space of Al has been dilated to 4.057 Å. Also marked in red are the positions of the ‘forbidden’ reflections (110), (210), (211), and (221) for

FCC Al, which are superlattice reflections of the L<sub>1</sub><sub>2</sub> structure. It can be seen that the measured peak positions are matched by L<sub>1</sub><sub>2</sub> structure reflections with lattice parameter 4.155 Å. It demonstrates that Yb, Er and Sc can be completely intersoluble to form L<sub>1</sub><sub>2</sub>, and the lattice parameter has been significantly reduced by adding Sc. Alloy 2D reflects the means of 5 reducing misfit by adding Mg to dilate the lattice space of Al and adding Sc to decrease the lattice parameter of L<sub>1</sub><sub>2</sub>. Thus Alloy 2D achieved the lowest misfit so far: 2.42% without losing the glass forming ability. Alloy 2D mainly consists of FCC Al and L<sub>1</sub><sub>2</sub> phases as devitrified microstructure.

10 **Example 6**

Alloy 1E in Table 5 was fabricated into ribbons using melt spinning technology (wheel speed 55 m/s). In the as melt-spun condition, Alloy 1E was characterized by X-ray diffraction. The diffractometer trace (Cu-K $\alpha$  radiation) of as-spun Alloy 1E in Figure 19 shows crystalline reflection peaks, indicating non-amorphous or partially non-amorphous state. It demonstrates 15 that TMs are necessary elements to promote glass in Al-TM-RE system. In addition, the ribbons are very brittle, also characteristic of the non-glassy state.

Figure 20 shows an X-ray diffractogram of Alloy 1E after devitrification at 425°C for 19 hours. This diffractogram shows that Alloy 1E only has two phases: fcc Al+L<sub>1</sub><sub>2</sub>. It also demonstrates that Yb, Er and Sc can be completely intersoluble to form L<sub>1</sub><sub>2</sub>, and the lattice 20 parameter has been significantly reduced by adding Sc. The misfit between FCC Al and L<sub>1</sub><sub>2</sub> is 3.03%.

The morphology of fcc+L<sub>1</sub><sub>2</sub> microstructure of Alloy 1E is shown in Figure 21. This SEM secondary electron image indicates that the particle size of L<sub>1</sub><sub>2</sub> precipitates is about 3 times larger than that of Alloy 1A, see Figure 9, although the misfit of Alloy 1A is a little bit 25 larger: 3.735%. It indicates that devitrification of glassy state can significantly reduce the particle size of precipitation in comparison with that by crystallization directly from fast cooling at the same rate.

In review, the range of composition processing parameters, microstructure characterization and physical properties is summarized in the following Table 7:

Table 7

	Composition				Processing Parameters			microstructure characterization (vol %)						Properties		
	Rare Earth Elemt.	Transition Metals and Mg, Li			glass	thermal	thermo-mechanical	amorphous		devitrified state						
(1)	Lata TM for glass fanning (2)	L12 modif- fer (3)	FCC elemt. For misfit (4)	at.%	RSP	aging temp. for thermal devitrifying	hot extrusion parameters	fcc	glass	others	fcc	L12	glass	others	room tempera- ture UTS	high tempera- ture UTS of 275- 410 Mpa
Maximum Ranges	Er,Lu, Yb,Tm U	Ni,Co, Fe,Cu	Sc,Y Ti,Zr, V,Cr, Mn,U, Nb	Mg,Zn, Ag	2%<sum(group (1))<16% 2%<sum(group (2))<7% sum(group (3))<5% sum(group (4))<7%	cooling rate >10 <sup>4</sup> °C /sec	>300°C		<15% bal.	<10% bal.	>10%	<16%	<20%	>500MPa	>250°C	
Preferred Ranges	Er,Lu, Yb,Tm	Ni,Co	Sc	Mg	5%<sum(group (1))<12% 2%<sum(group (2))<7% Sc<4% Mg<8%	cooling rate >10 <sup>4</sup> °C /sec	>325°C <475°C	>350°C <450°C 5,9,11:1 ratio	<8% bal.	<3% bal.	>20- 25%	<8%	<15%	>800MPa	>275°C	
Optimum Ranges	Er,Yb;	Ni,Co	Sc		7%<sum(Er,Yb,S c)<8% 3%<sum(Ni,Co)< 6%	>10 <sup>5</sup> °C /sec, atomize d powder with >325 mesh	>370°C <450°C	400°C 11:1 ratio	<6% bal.	<1% bal.	>30%	<3%	<5%	>1000MPa	>300°C	
Examples	Er,Yb	Ni,Co	Sc,Y	Mg	7%<sum(Er,Yb)< 8% sum(Ni,Co)<5.5 % sum(Sc,Y)<5%, Mg<3%	about 10 <sup>6</sup> °C /sec	425°C	400°C 11:1 ratio	100 % or <20 %	bal.	>28% <32%	<3%	<15- 20%	480VHN (1A) which is > 1000MPa		
					Preferred Methods: Atomization and consolidation with extrusion				"							

Variations of the described aluminum alloy as well as the process for manufacture thereof and the product created by the process are available to provide the expected functionality of high short-term and long-term strength at temperatures above about 300°C. Thus the invention is to be limited only by the following claims and equivalents thereof.

## CLAIMS

What is claimed is:

1. An aluminum alloy characterized by high strength including high strength in the temperature range greater than about 250°C comprising, in combination:

5 an alloy mixture in primarily crystalline form having at least about 30% by volume fcc phase and at least about 10% by volume L<sub>1</sub><sub>2</sub> ductile precipitate phase, said alloy consisting essentially of at least one transition metal selected from the group consisting of about 2 to 12 atomic percent Ni, Co, Ti, Fe, Y, Sc, Cu, Zn, V, Cr, Mn and Li, and at least one rare earth material selected from the group consisting of about 2 to 10 15 atomic percent Er, Tm, Yb, Lu, and the balance Al.

2. The alloy of claim 1 further including Mg in an amount up to about 5% atomic fraction.

3. The alloy of claim 1 having a tensile strength of at least about 275Mpa at 250°C.

4. A method for manufacture of an aluminum alloy characterized by high strength including high strength in a temperature range greater than about 250°C comprising the steps

15 of:

(a) forming a melt mixture of aluminum, about 2 to about 12 atomic percent of at least one metal selected from the group consisting of Ni, Co, Ti, Fe, Y, Sc, Cu, Zn, V, Cr, Mn and Li, and at least one rare earth selected from the group consisting of about 2 to 15 atomic percent Er, Tm, Yb, and Lu;

20 (b) converting said melt to an amorphous state wherein the mixture comprises at least about 70% by volume amorphous material; and

(c) devitrifying said amorphous material to at least in part a crystalline matrix of at least about 30% by volume fcc and at least about 10% by volume L<sub>1</sub><sub>2</sub> phase.

5. The method of claim 4 wherein the step of converting the melt comprises cooling in the 25 range of at least about 10<sup>3</sup>°C per second.

6. The method of claim 4 wherein the step of devitrification is selected from a group consisting of thermoprocessing, thermomechanical processing and combinations thereof. A high temperature and high strength aluminum-based alloy processed through a primarily greater than about 70% in volume amorphous state and then devitrified into greater than about 70% in 30 volume crystalline microstructure with fcc matrix and strengthening by about a ductile L<sub>1</sub><sub>2</sub> precipitate phase greater than about 10% L<sub>1</sub><sub>2</sub>, and fcc greater than about 30% in volume.

7. The method of claim 4 wherein converting the melt comprises a step selected from the group consisting of gas powder atomization, water powder atomization, melt spinning, spray casting and combinations thereof.

8. The method of claim 4 wherein the step of devitrifying amorphous material comprises a 5 step selected from the group consisting of hot isostatic pressing, thermal aging, extrusion and combinations thereof.

9. An aluminum alloy characterized by high strength including high strength at a temperature greater than about 250°C made by a process comprising the steps of:

(a) formulating a melt comprised of Al; at least one transition metal (TM) 10 selected from the group consisting of Ni, Co, Ti, Fe, Y, Sc, Cu, Zn, V, Cr, Mn, Li and Mg; and at least one rare earth selected from the group consisting of Er, Tm, Yb, Lu;

(b) converting the melt to at least about 70% by volume amorphous material; and

(c) devitrifying at least in part the amorphous material to a mixture of fcc and 15 L<sub>1</sub><sub>2</sub> crystalline precipitate phase material.

10. The alloy product by the process of claim 9 wherein the transition metal is provided in an amount of about 2 to 10 atomic percent.

11. The alloy product by the process of claim 9 wherein the rare earth material is provided in an amount of about 2 to 10 atomic percent.

20 12. The alloy product by the process of claim 9 further including an additive of Mg to the melt.

13. The alloy product by the process of claim 9 wherein devitrifying the amorphous material comprises forming at least about 30% by volume fcc material.

25 14. The alloy product by the process of claim 9 wherein devitrifying the amorphous material comprises forming at least 10% by volume L<sub>1</sub><sub>2</sub> material.

15. The alloy product by the process of claim 9 wherein converting the melt to amorphous material comprises at least one step selected from the group consisting of gas powder atomization, water powder atomization and melt spinning.

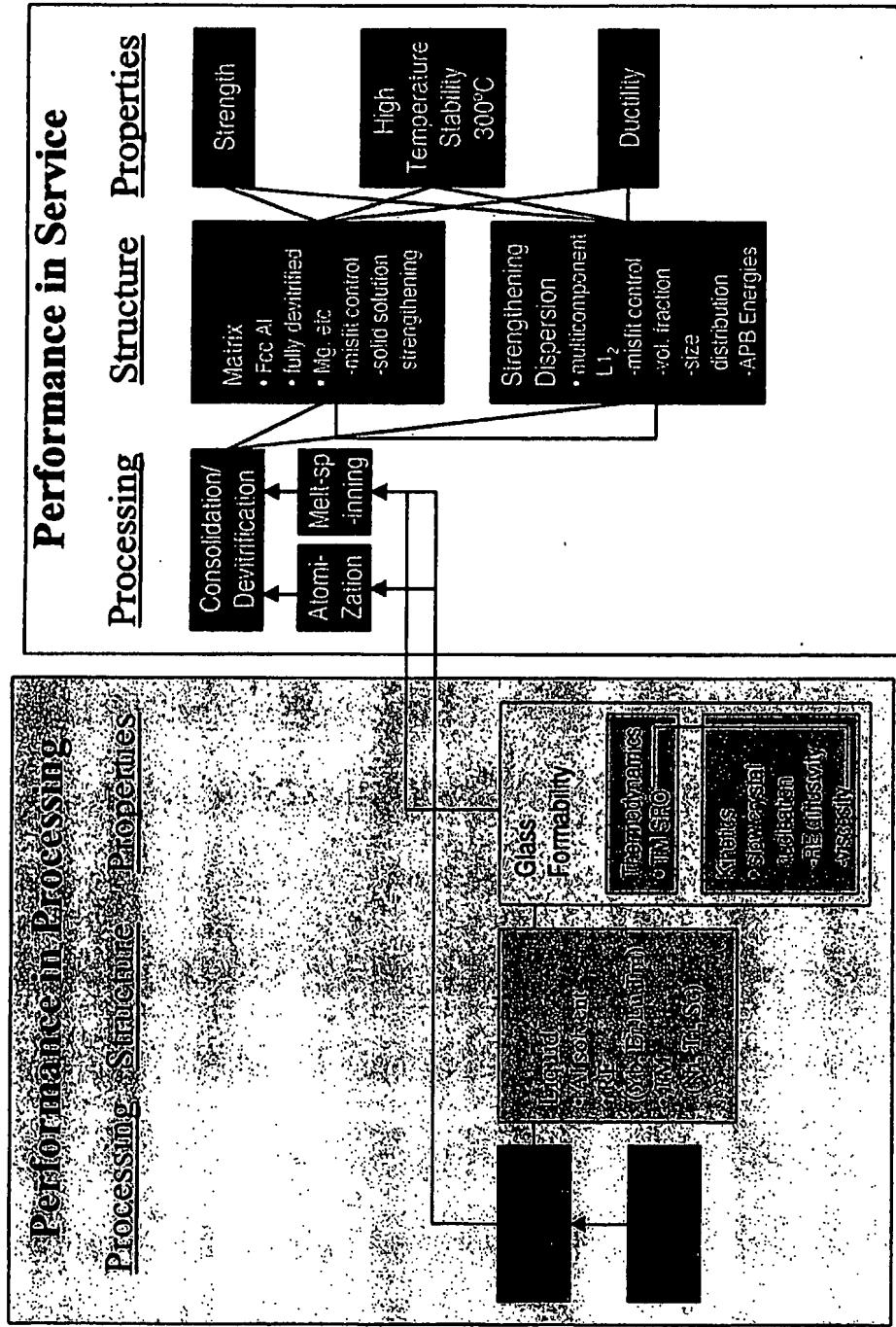
30 16. The alloy product by the process of claim 9 wherein devitrification comprises at least one step selected from the group consisting of hot isostatic pressing, thermal aging, and extrusion.

17. The method of claim 4 wherein converting the melt comprises the step of rapid solidification processing.

18. The product by the process of claim 9 wherein converting the melt comprises rapid solidification processing.

19. The alloy of claim 1 or claim 9 wherein the precipitate phase has a grain size diameter in the range of less than about 80 nm.

# *Damage Tolerant Amorphous Metal Alloys*



**Figure 1**

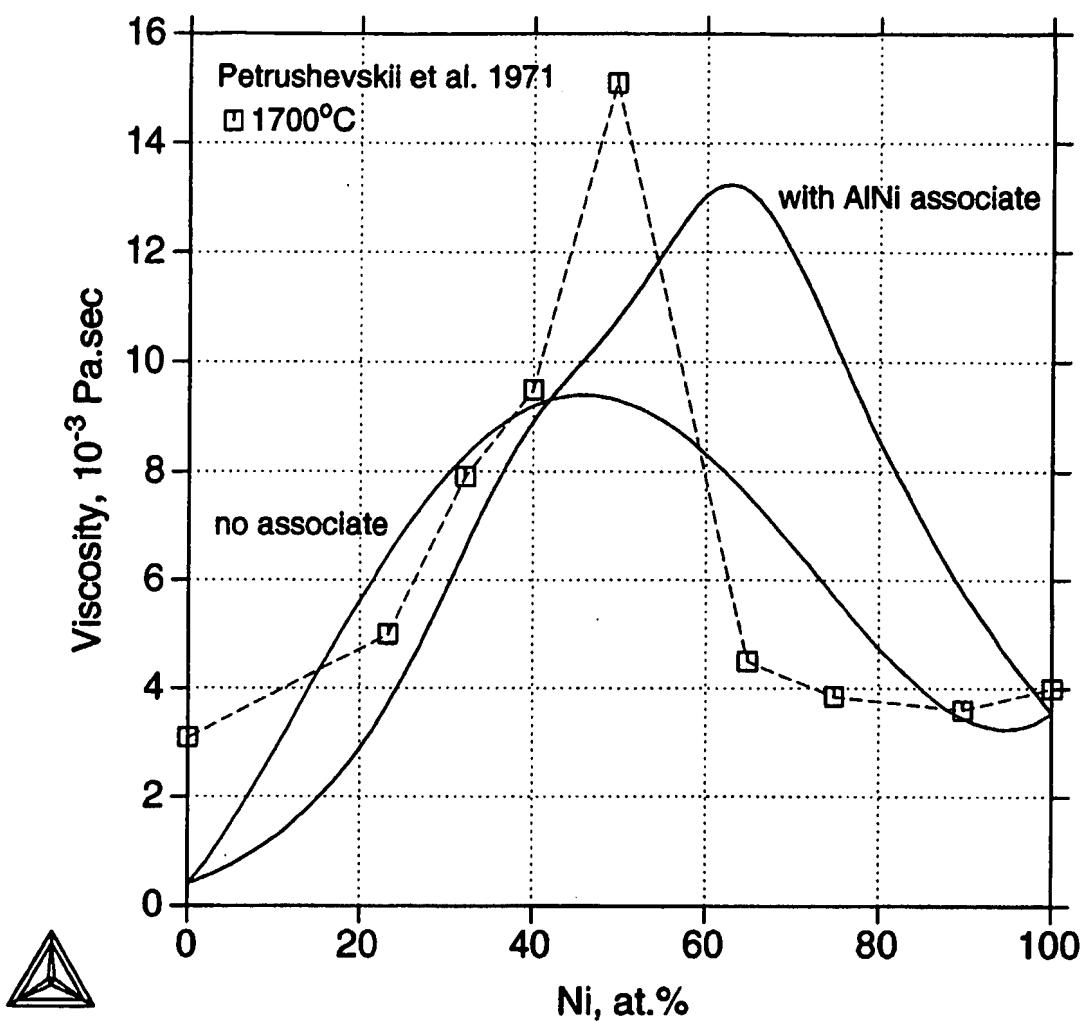


Figure 2

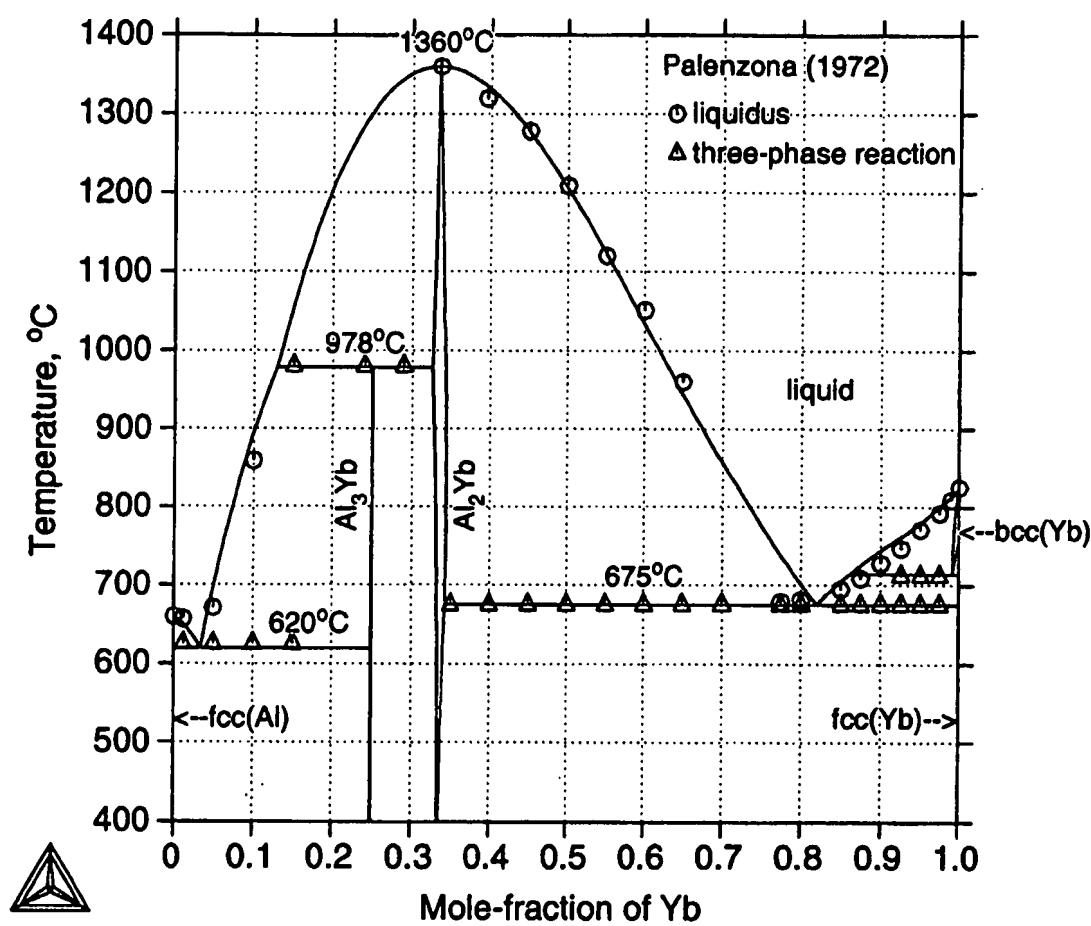


Figure 3

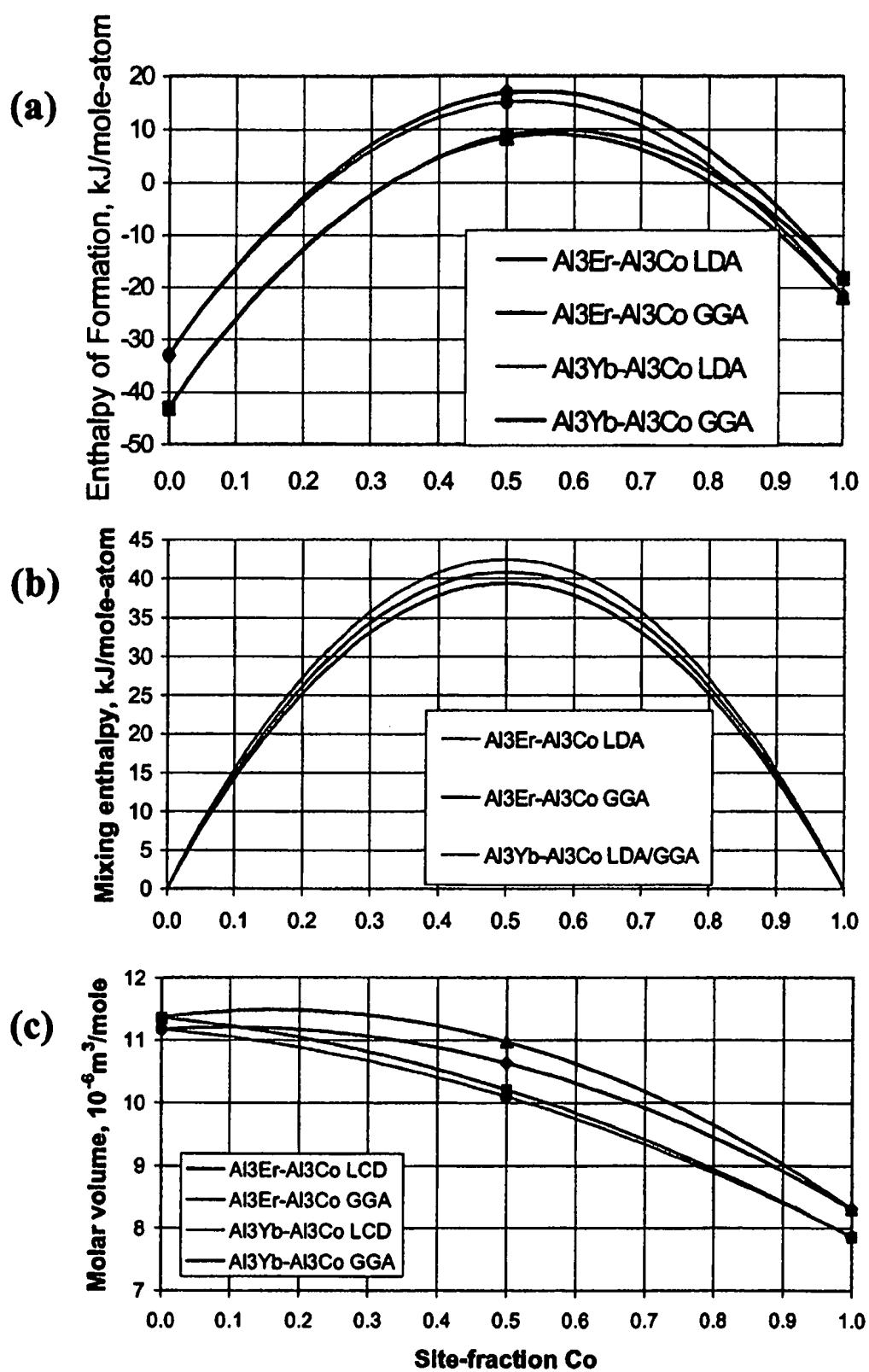
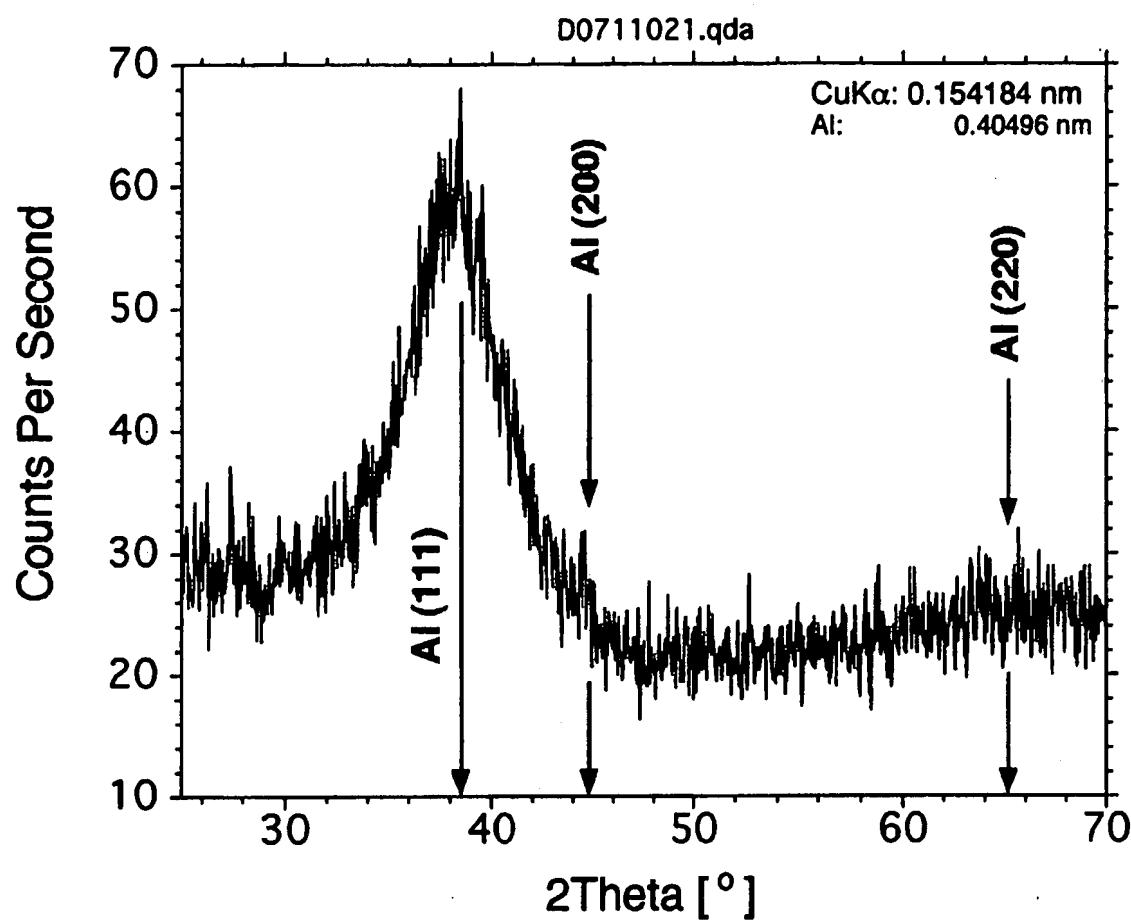


Figure 4

**1A Al-Ni-Co-Yb-Er as melt-spun 55 m/s****Figure 5**

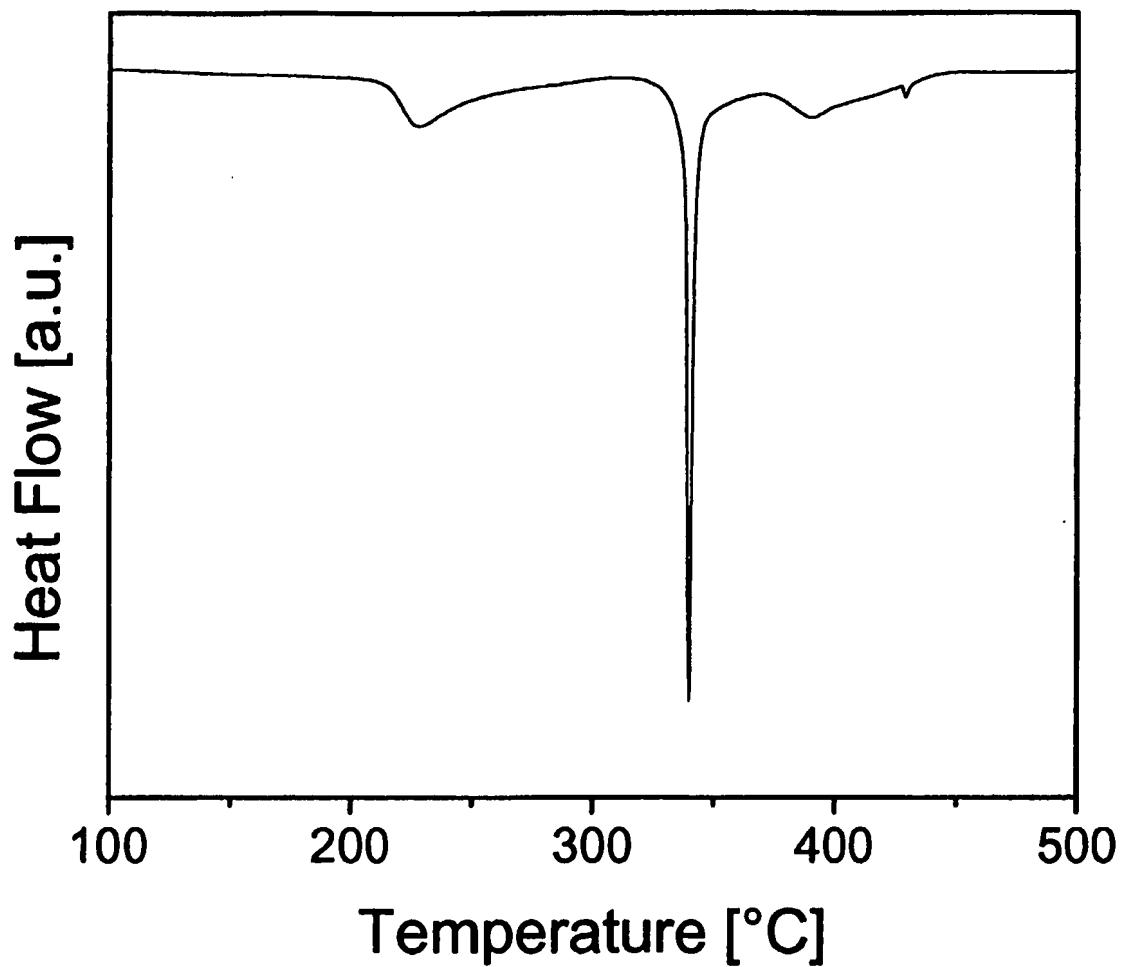


Figure 6

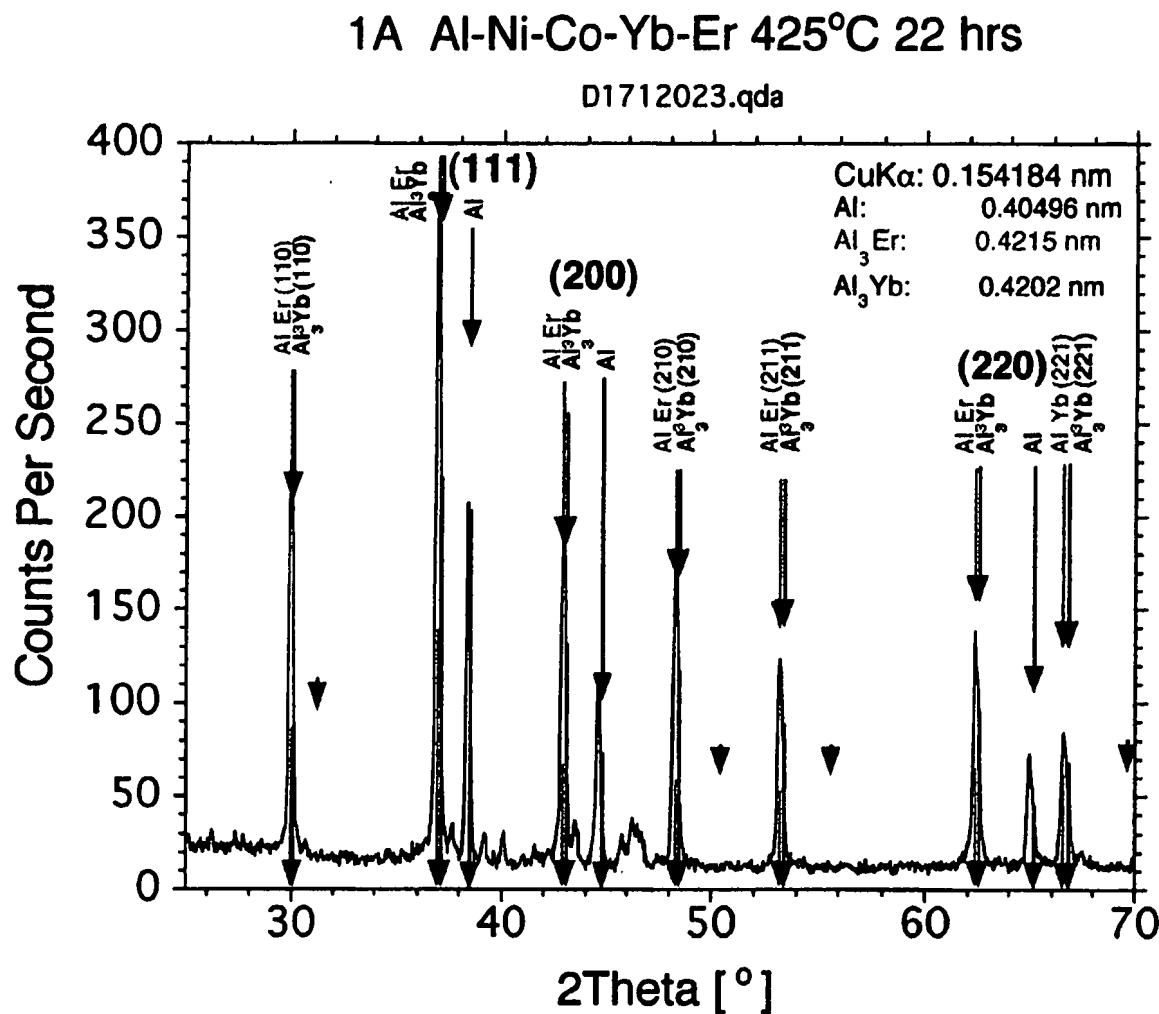


Figure 7

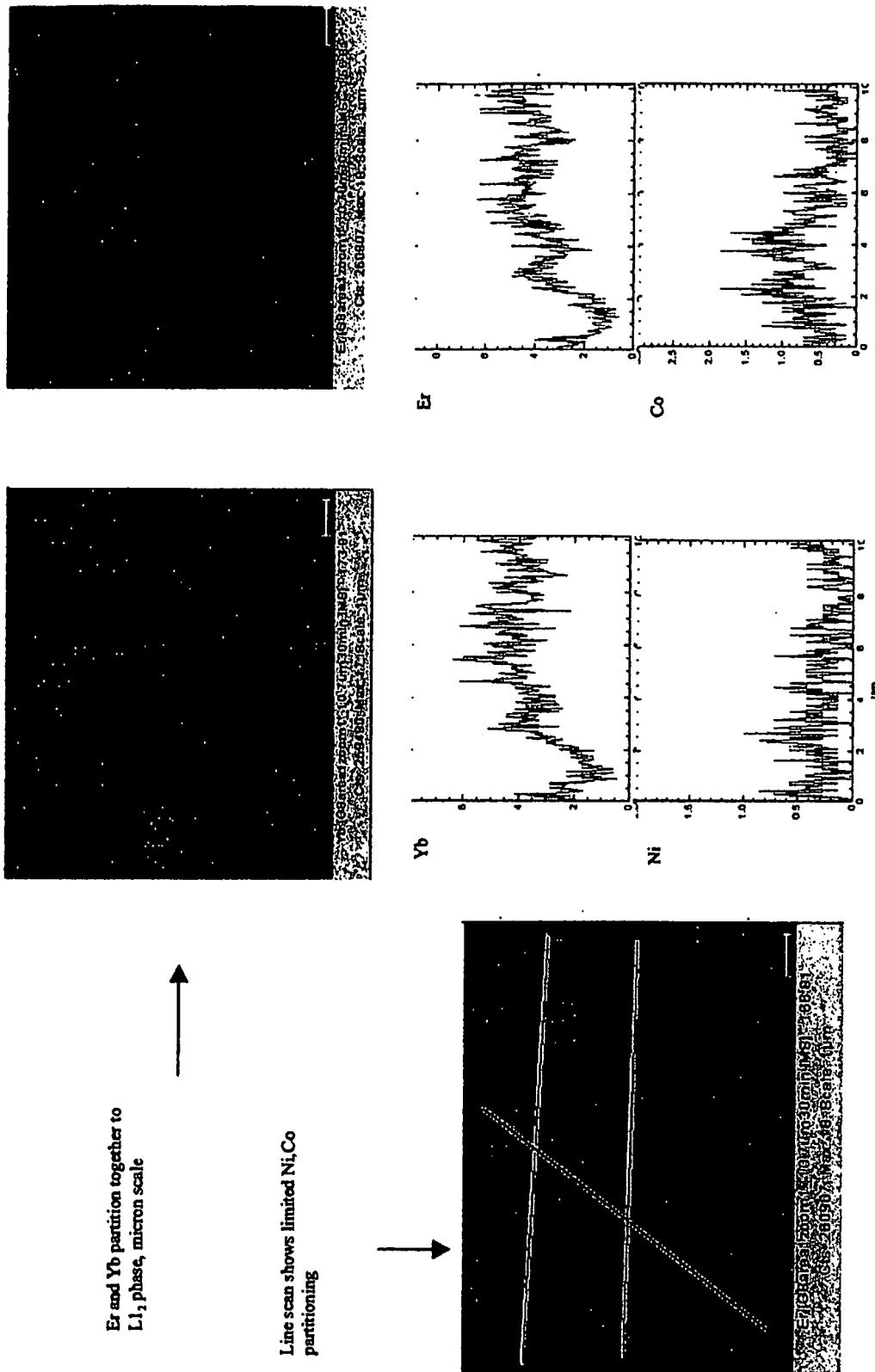


Figure 8

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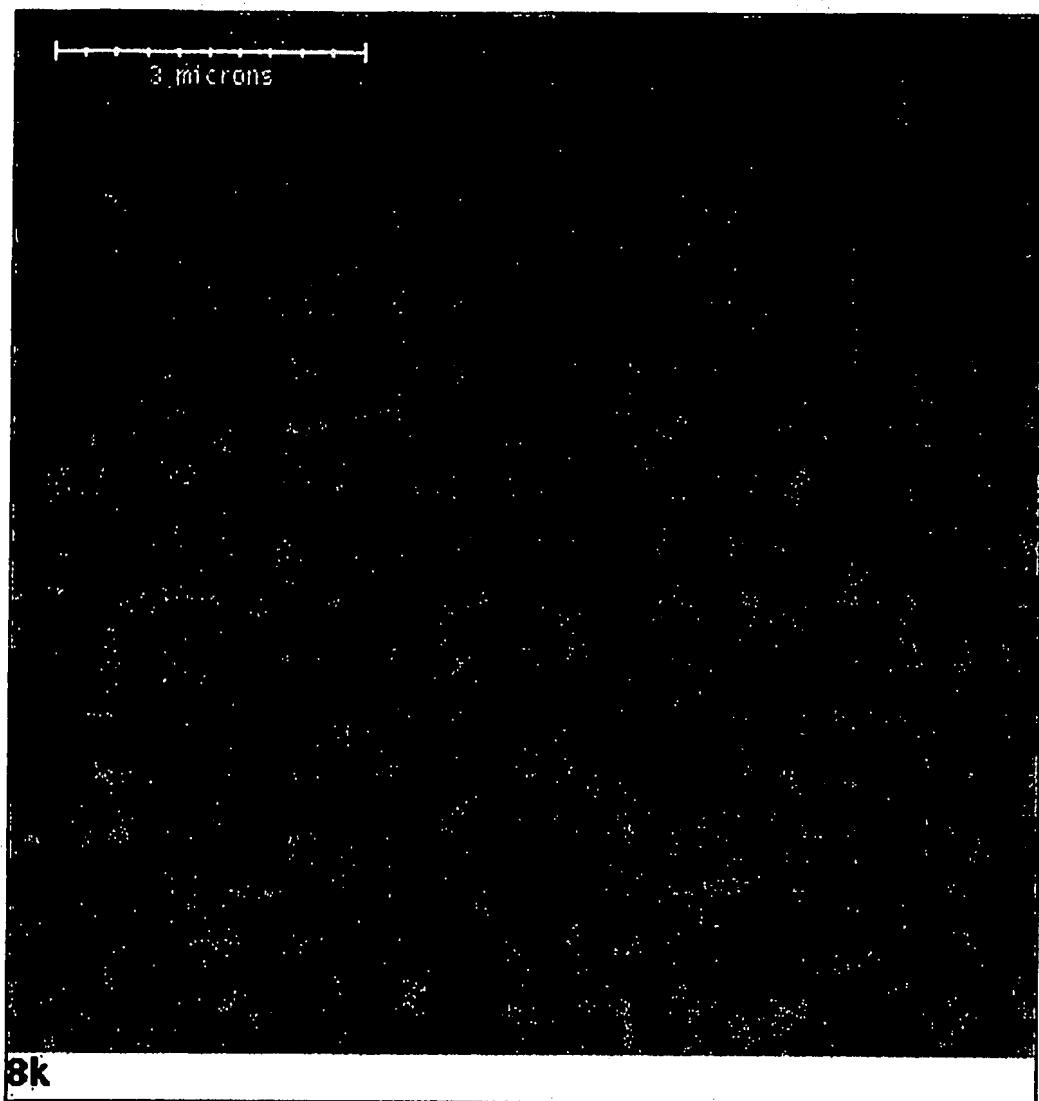


Figure 9

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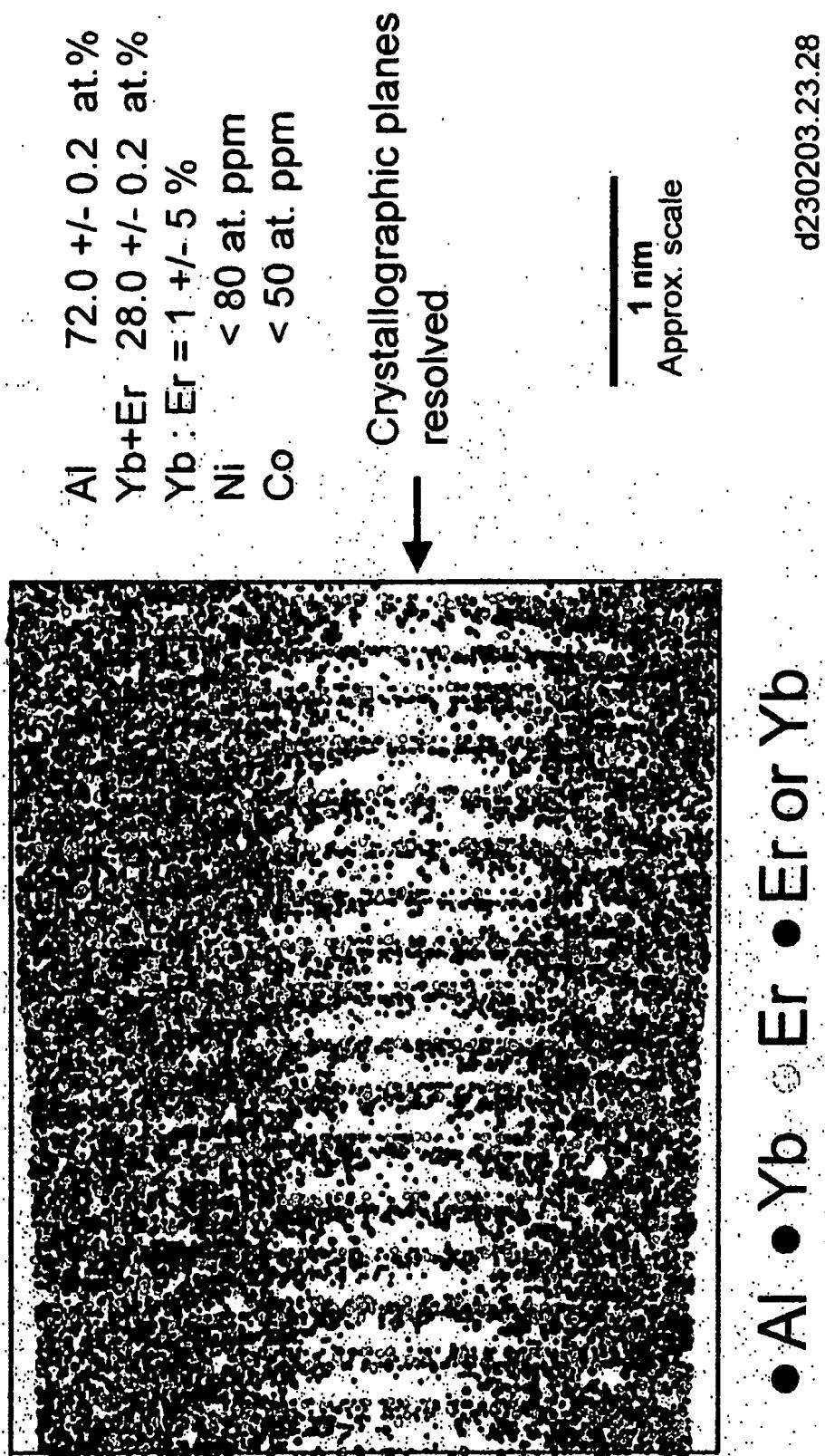


Figure 10

11/21

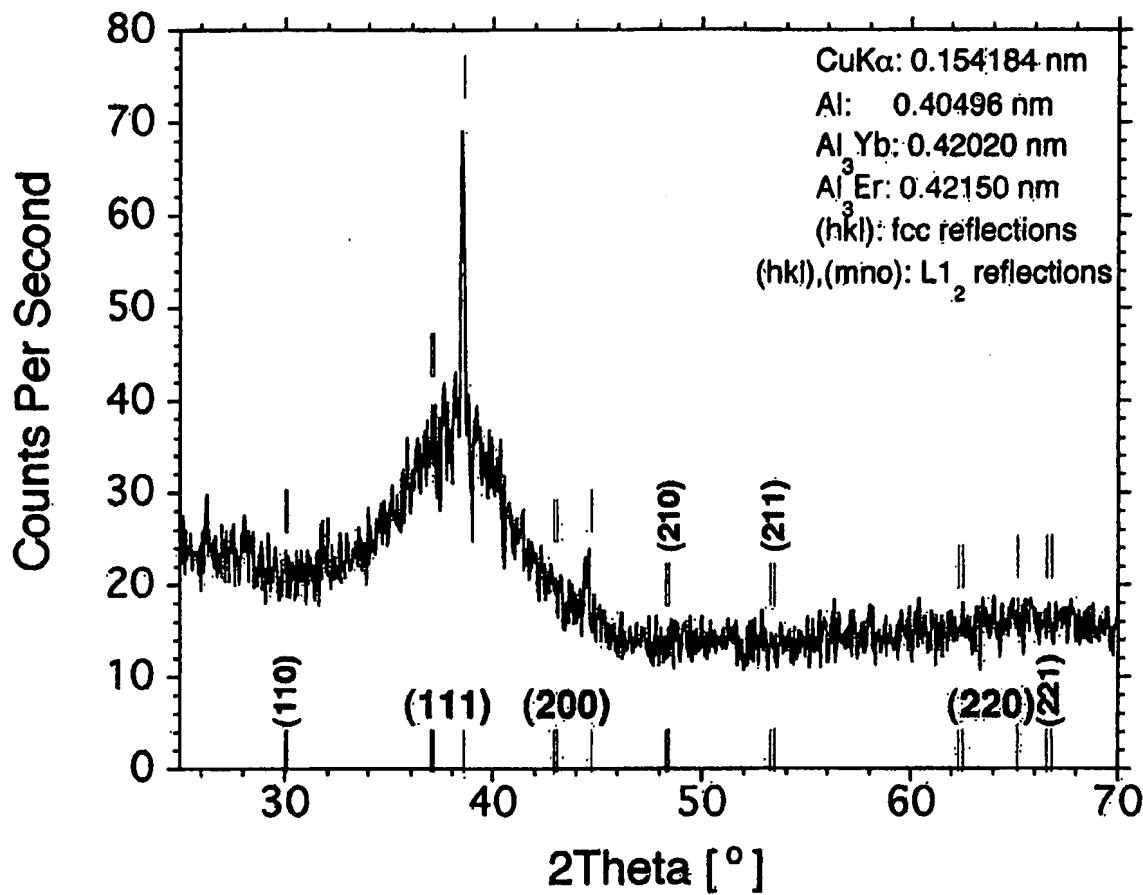


Figure 11

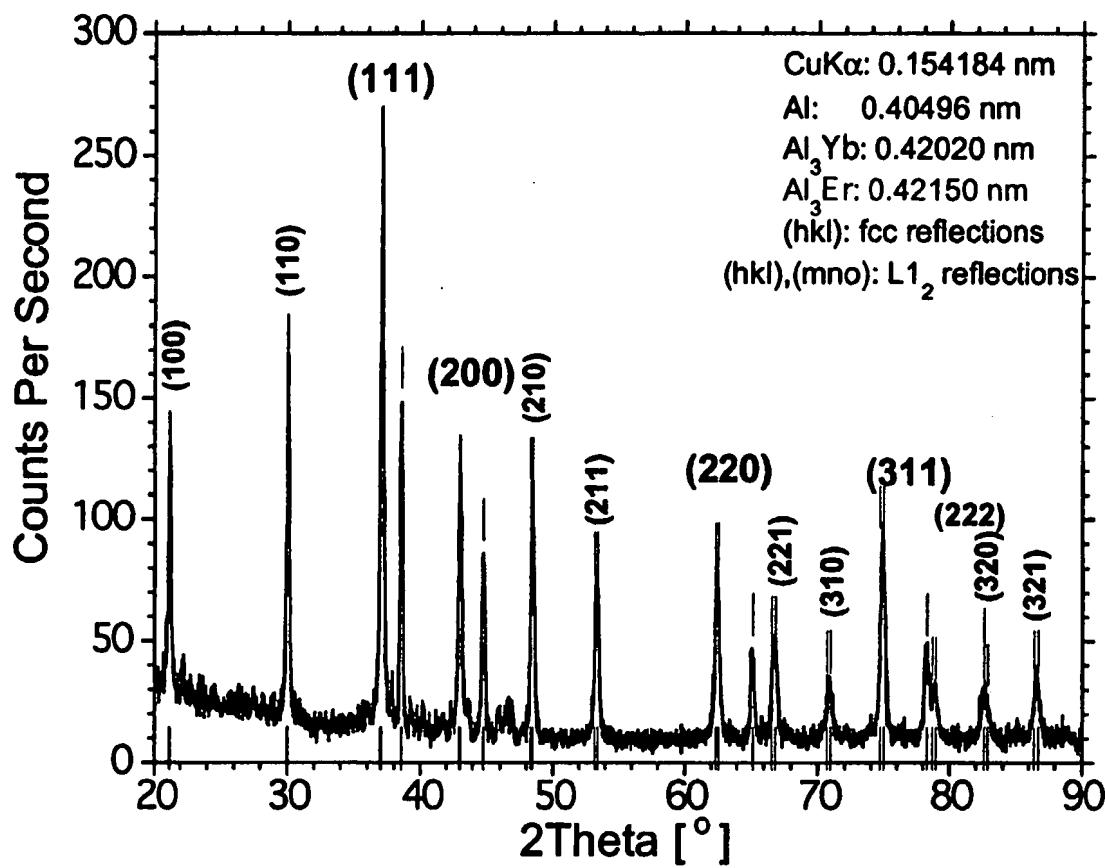


Figure 12

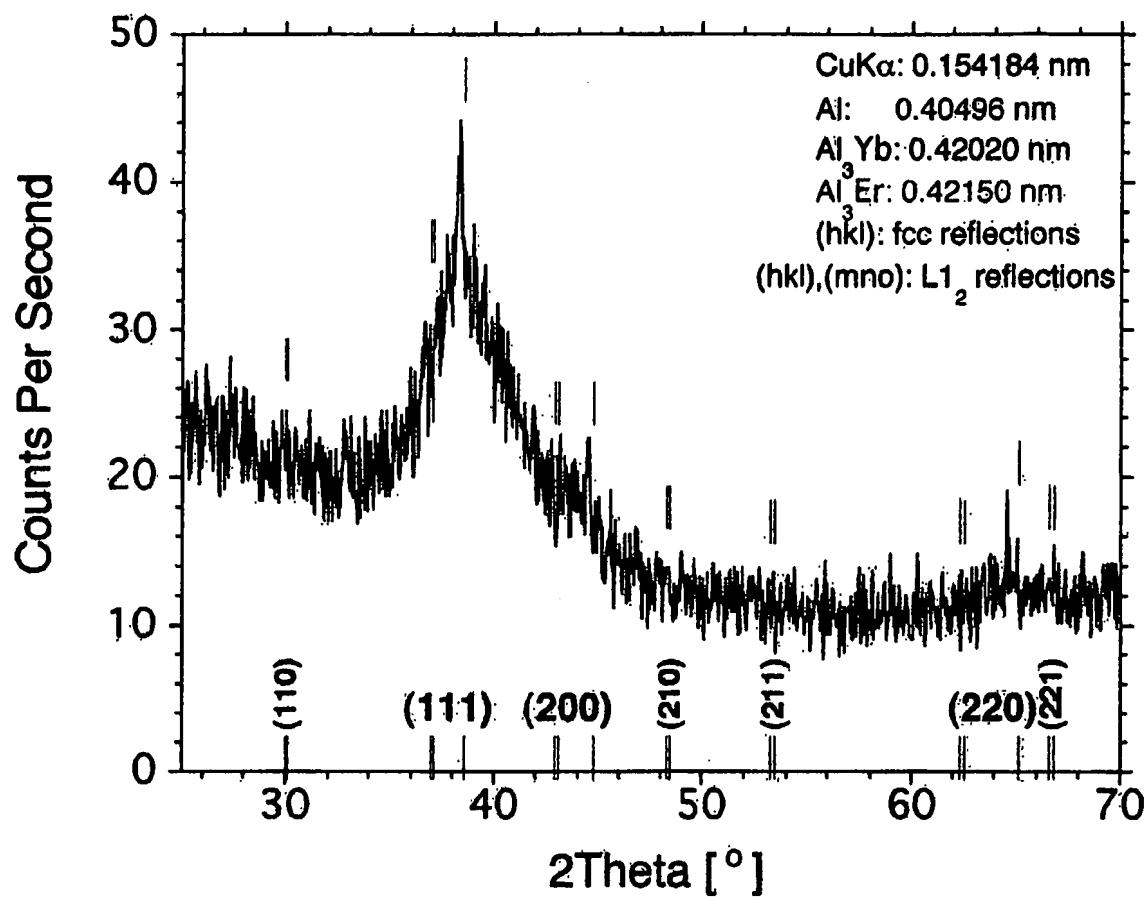


Figure 13

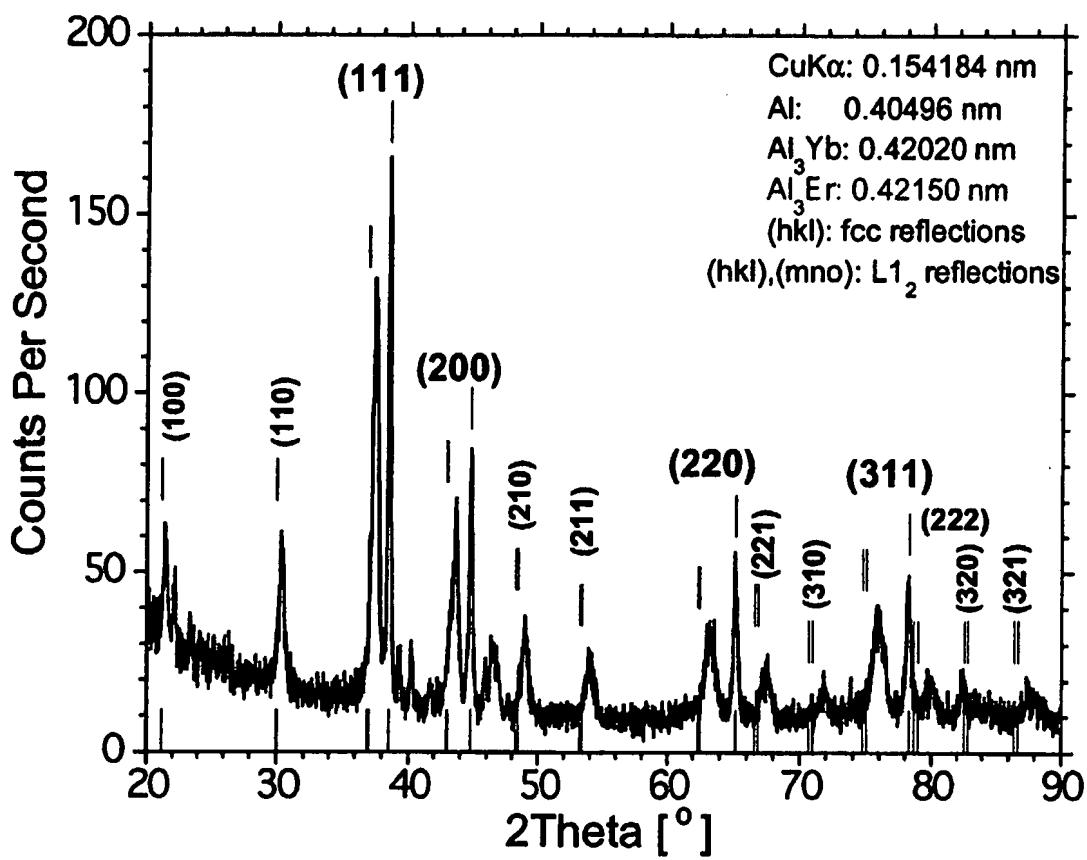


Figure 14

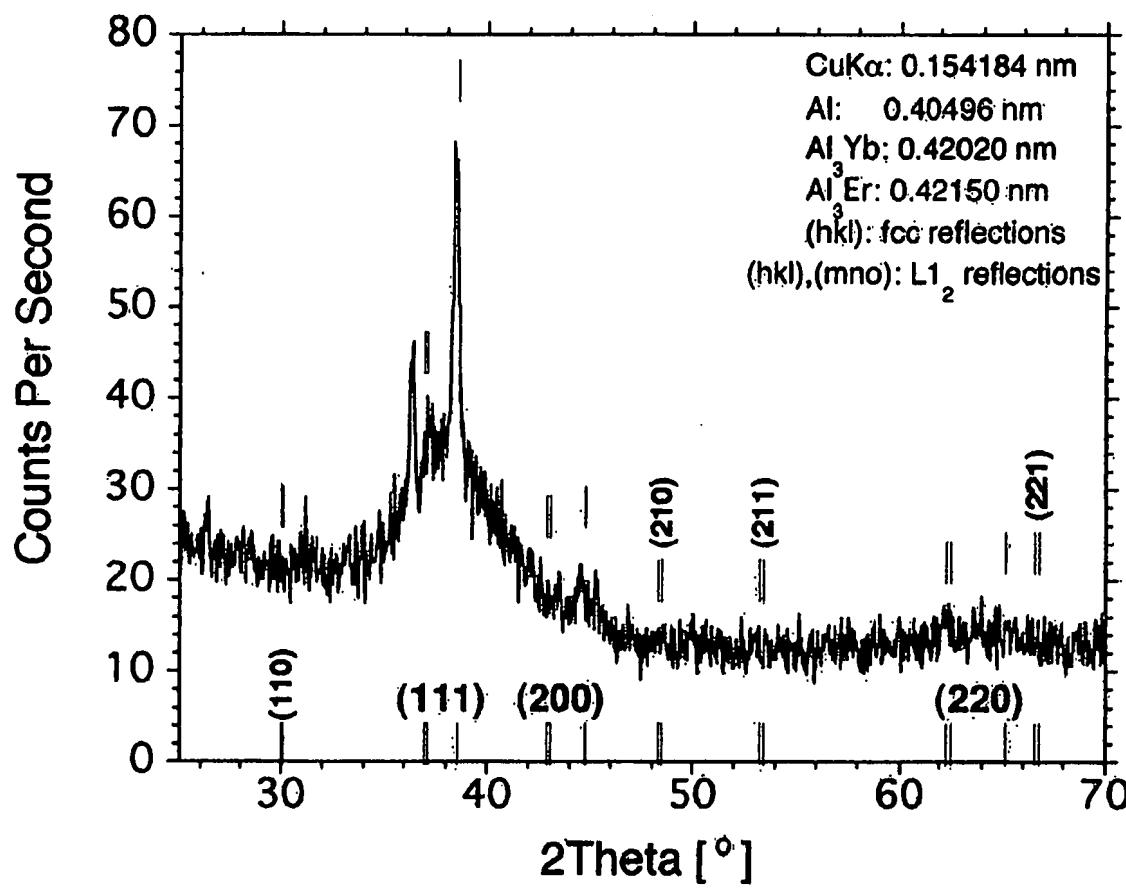


Figure 15

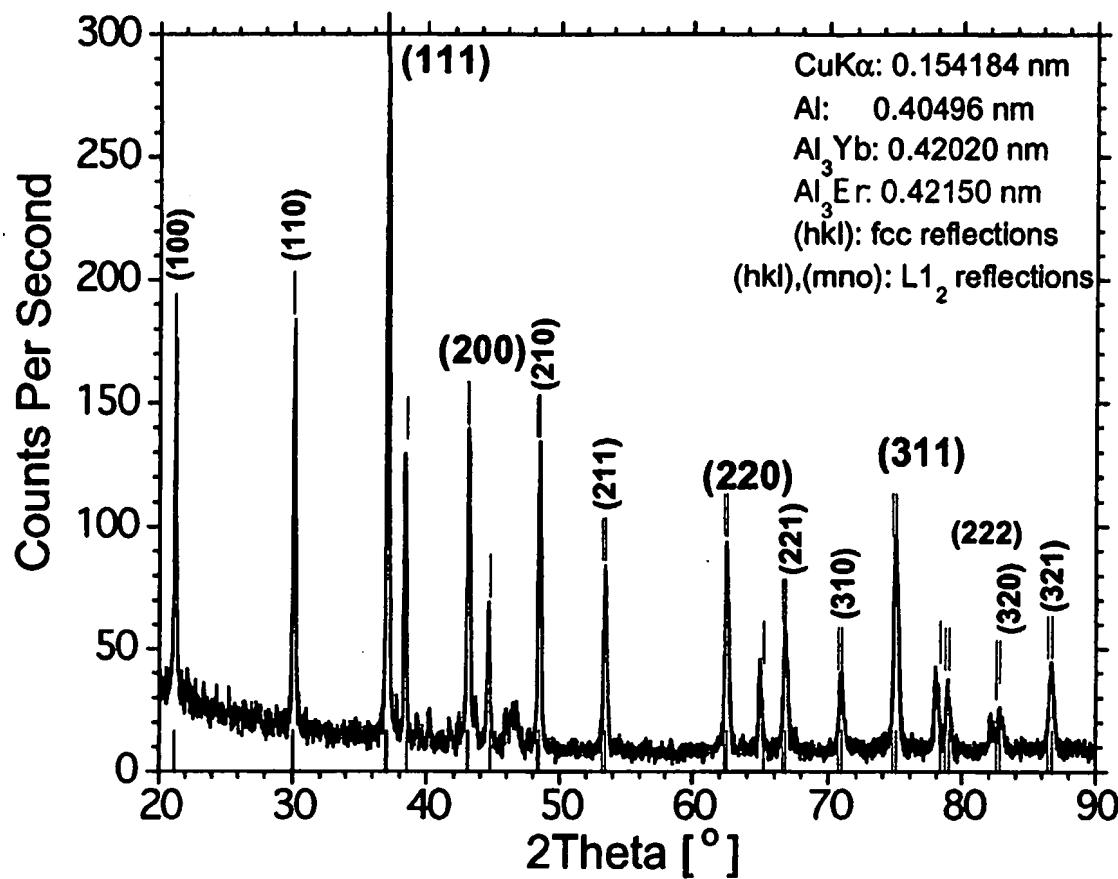


Figure 16

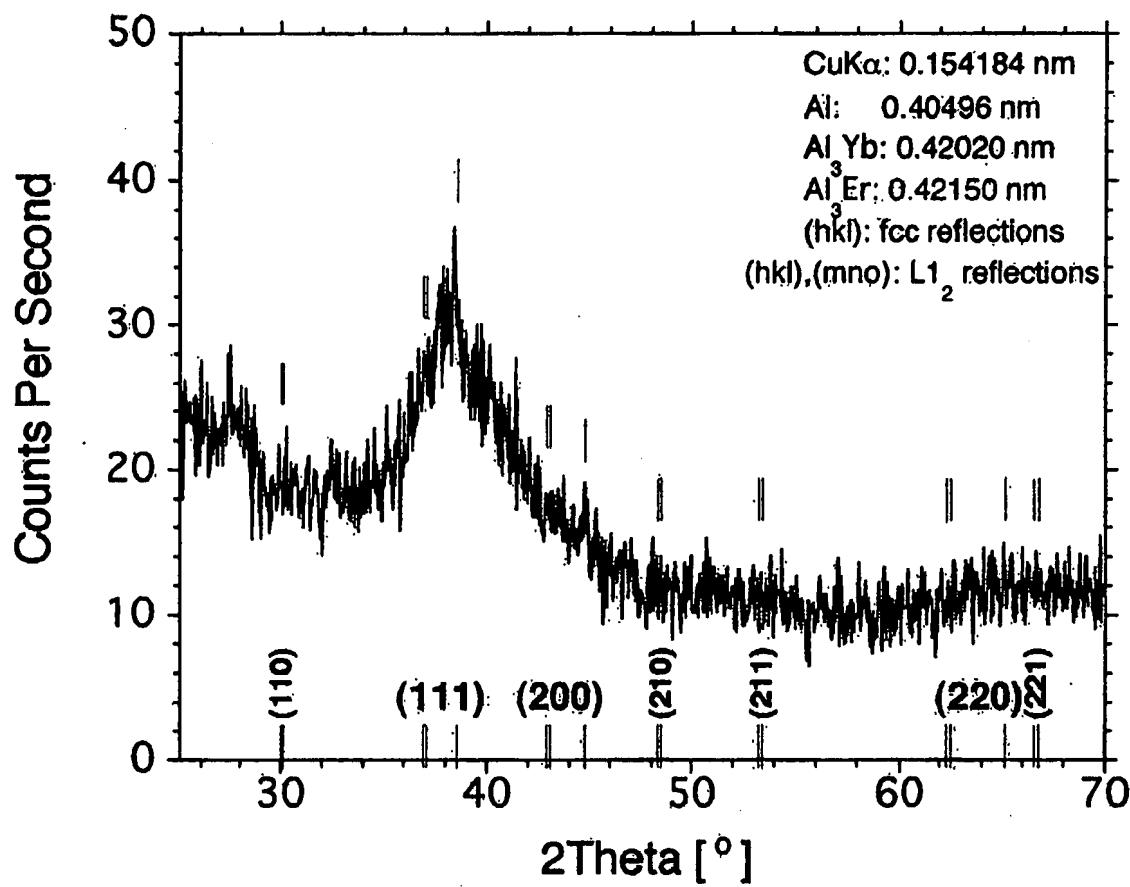


Figure 17

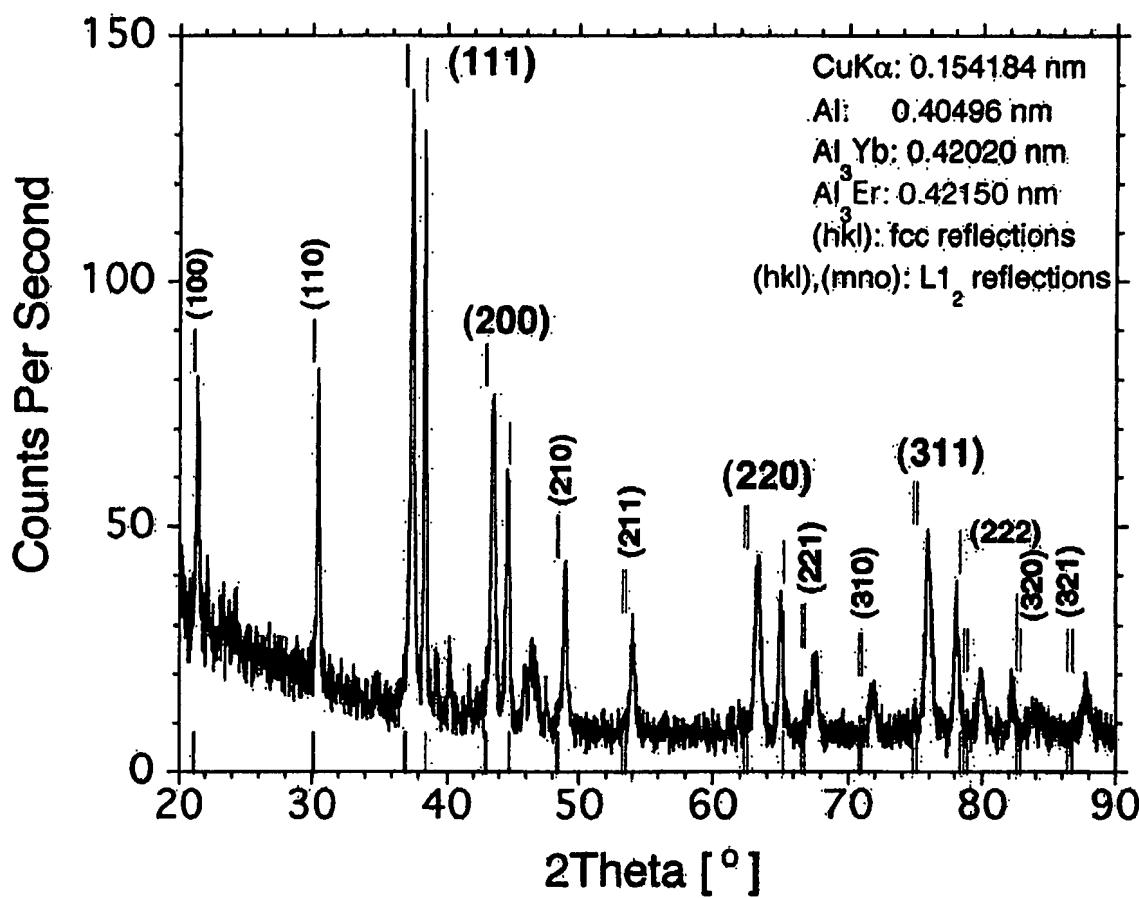


Figure 18

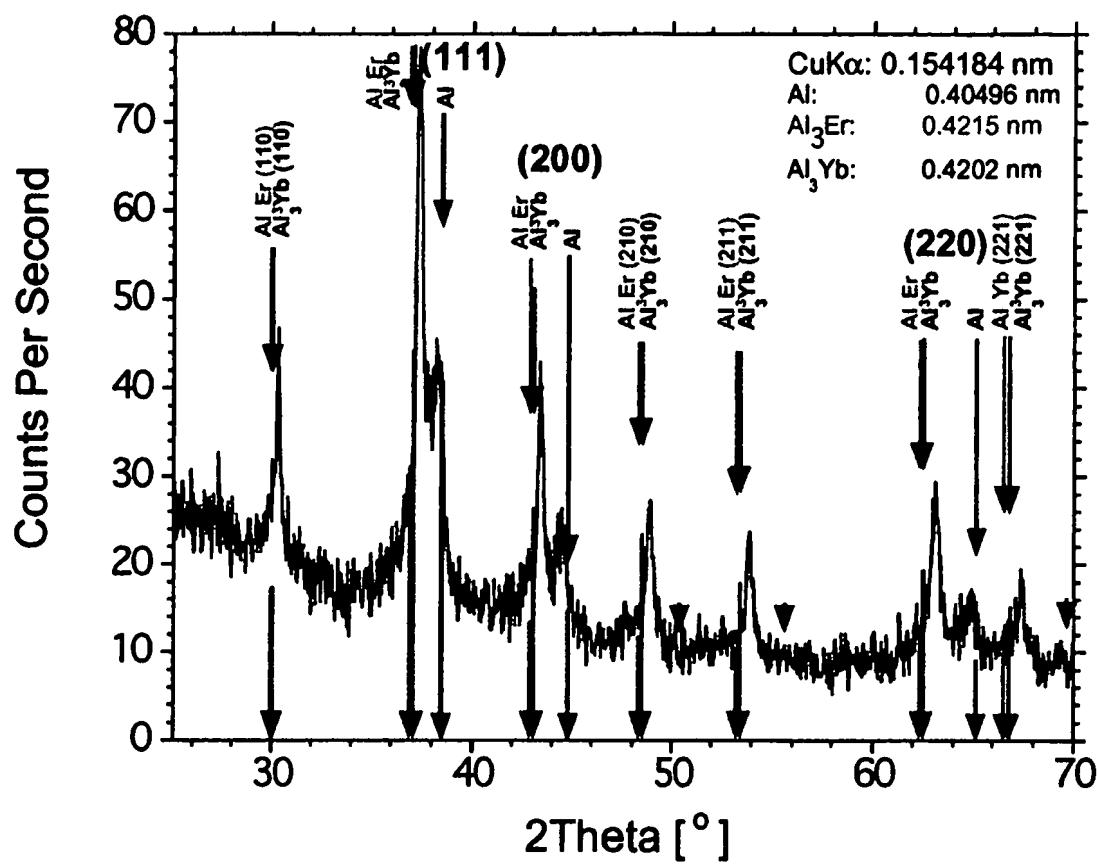


Figure 19

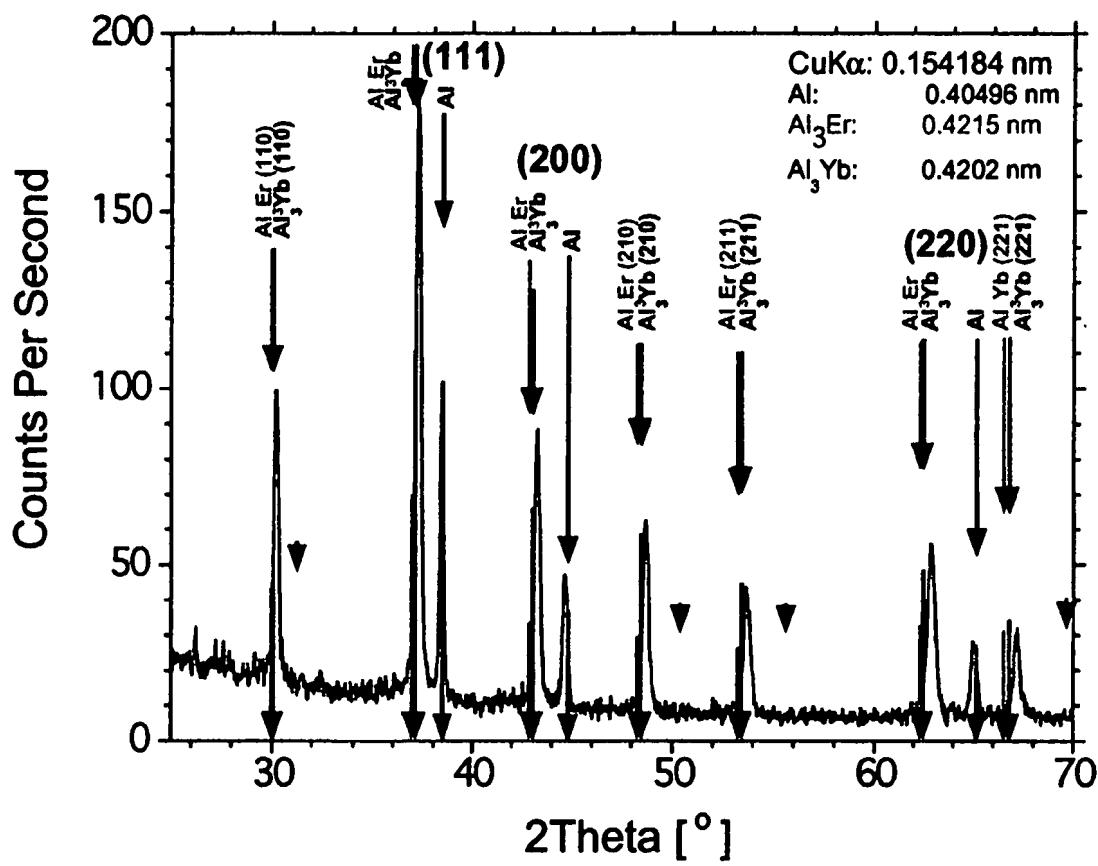


Figure 20

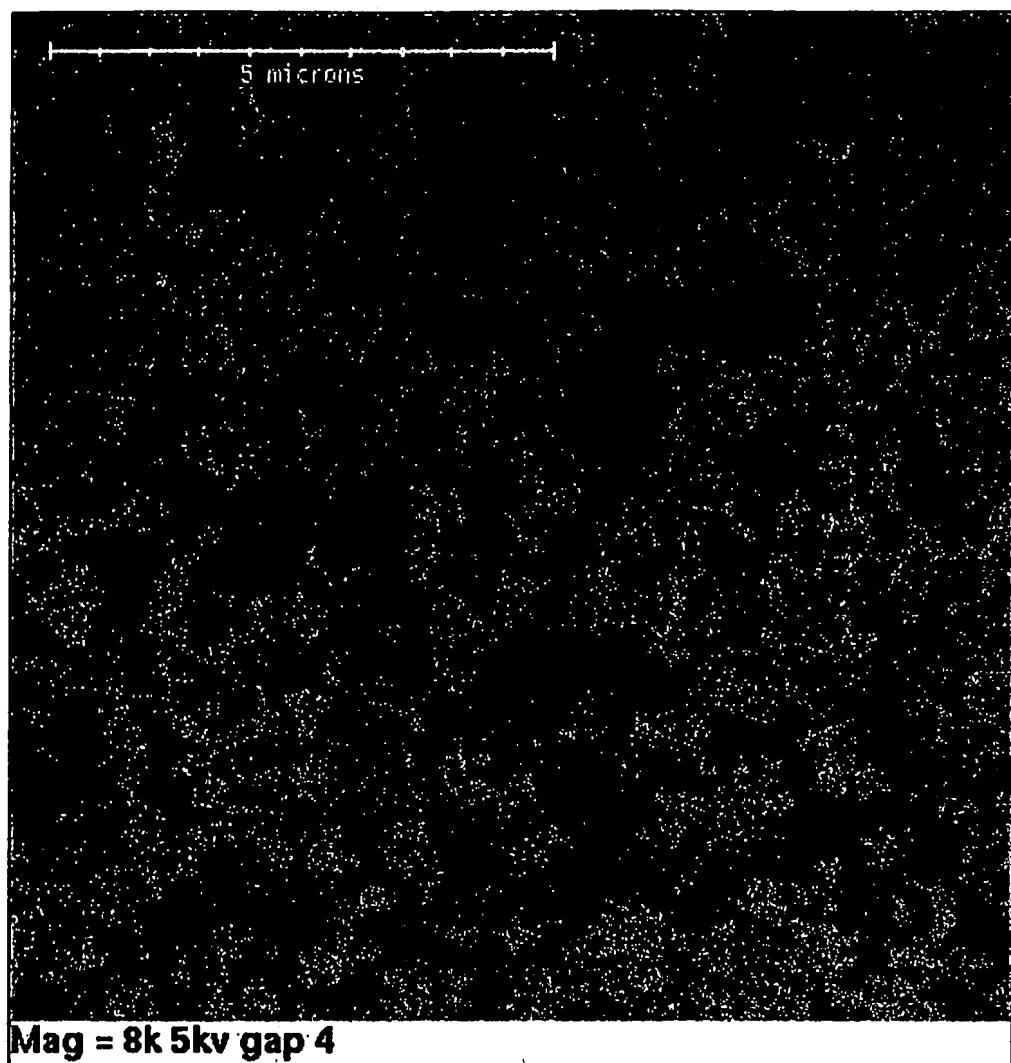


Figure 21